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FLASHBACK FLAME ARRESTER DEVICES FOR FUEL CARGO TANK VAPOR VENTS

R. A. BJORKLUND R. O. KUSHIDA



FINAL REPORT MARCH 1931



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PREFACE

The work described in this report was jointly sponsored by the DOT/U.S. Coast Guard, Marine Technology Division, Office of Research and Development, and NASA/Office of Space and Terrestrial Application, Technology Transfer Division, and was performed by the Control and Energy Conversion Division, Propulsion Systems Section, under the program cornizance of the Office of Energy and Technology Applications of the Jet Propulsion Laboratory.

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CONTENTS

ľ.	SUMMA	RY	1-1
II.	INTRO	DUCTION	2-1
III.	TEST	FACILITY DESCRIPTION	3-1
	Α.	GENERAL	3-1
	g.	AIR COMPRESSOR SYSTEM	3-1
	c.	FUEL SYSTEM	3-1
	D.	FUEL VAPORIZER AND CONDENSER LOOP	
	E.	FUEL AND AIR INDUCTION SYSTEM	3-5
	F.	FACILITY PIPING	3-5
	G.	FLAME TEST CHAMBER	3-6
	н.	HYDROGEN/AIR SPARK IGNITER	3 - 8
	I.	EXHAUST-BURN STACK	3 - 8
	J.	SUSTAINED BURNING TEST FACILITY	3 ~ 8
IV.	INSTR	RUMENTATION AND CONTROLS	h-1
	Α.	GENERAL DESCRIPTION	4-1
	В.	STEADY-STATE DATA	4-1
	C.	TRANSIENT-STATE DATA	4-4
	D.	GAS-SAMPLE ANALYSIS SYSTEM	4-7
	E.	PHOTOGRAPHIC DATA	4-7
	F.	PARAMETER MEASUREMENT AND CALCULATION UNCERTAINTIES	4-9
٧.	TEST	OPERATING PROCEDURES	5 - 1
	Α.	GENERAL SAFFTY REQUIREMENTS	5-1
	В.	CPERATING PROCEDURE CHECK LISTS	5 - 1

	1.	Pretest System Checkouts	5-1
	2.	Fuel Transfer Procedures	5-2
	3.	Test Preparations	5-2
	4.	Blockhouse Preparation	5-3
	5.	Countdown	5 - 3
	6.	Posttest	5-4
VI.	FACIL	ITY CHECKOUT TESTS	6-1
	Α.	SUBSCALE FLAME CHAMBER TESTS	6-1
	В.	FULL-SCALE FLAME CHAMBER TESTS	6 - 2
VII.	DESCR	IPTION OF FLAME ARRESTER TEST ASSEMBLIES	7-1
	Α.	GENERAL	7-1
	В.	SINGLE 30-MESH SCREEN ARRESTER	7-1
	C.	DUAL 20-MESH SCREEN ARRESTER	7-2
	D.	SPIRAL-WOUND, CRIMPED METAL RIBBON ARRESTER	7-2
	Ε.	PACKED BED OF BALLAST RINGS ARRESTER	7 - 5
VIII.	FLASH	BACK FLAME ARRESTER TESTS	8-1
	Α.	TEST PROGRAM LOGIC	8-1
	В.	PROPANE/AIR MIXTURE SCREENING TESTS	8-3
	C.	ETHYLENE/AIR MIXTURE SCREENING TESTS	8-5
	D.	GASOLINE/AIR MIXTURE TESTS	8-7
	E.	METHANOL/AIR MIXTURE TESTS	8-10
	F.	TOLUENE/AIR MIXTURE TESTS	8-11
	G.	DIETHYL ETHER/AIR MIXTURE TESTS	8-12
	н.	BUTANE/AIR MIXTURE TESTS	8-13
	I.	ACETALDEHYDE/AIR MIXTURE TESTS	8-13
	J.	ARRESTER SELECTION FOR SUSTAINED BURNING TESTS	8-14

IX.	SUSTA	INED BURNING ARRESTER TESTS	9-1
	Α.	PROPANE/AIR MIXTURE TESTS	9-1
	1.	Single 30-Mesh Screen Arrester, 15.2-cm Diameter	9-1
	2.	Dual 20-Mesh Screen Arrester, 15.2-cm Diameter	9-1
	3•	Single 30-Mesh Screen Arrester, 25.4-cm Diameter	9 - 2
	4.	Dual 20-Mesh Screen Arrester, 25.4-cm Diameter	9 - 6
	5.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester	9 - 8
	6.	Packed Bed of Aluminum Ballast Rings Arrester	9-10
	В.	ETHYLENE/AIR MIXTURE TESTS	9-1
	1.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester	9-13
	2.	Packed Bed of Aluminum Ballast Rings Arrester	9-15
х.	CONCL	USIONS	10-1
XI.	RECOM	MENDATIONS	11-
REFER	ENCES		12-1
APPEN	DIXES		
	Α.	TEST CONFIGURATION LOG	A-1
	В.	TABULAR SUMMARY OF STEADY-STATE MEASURED AIR AND FUEL SYSTEM TEST CONDITIONS	B-1
	C.	TABULAR SUMMARY OF TRANSIENT-STATE MEASURED FLAME SPEED AND PEAK PRESSURE RISE TEST DATA	C-1
	D.	TABULAR SUMMARY OF AVERAGED MEASURED FLAME SPEED AND PEAK PRESSURE RISE FOR ALL FUELS	D-1
	Ε.	TABULAR SUMMARY OF TEMPERATURE MEASUREMENTS FOR SUSTAINED BURNING TESTS	E-1

FIGURES

1-1.	B-Stand Facility, Edwards Test Station	1-2
2-1.	A Flammable Fuel/Air Mixture Flowing Slowly Out of a Vent Stack Into the Open Air	2-4
2-2.	An External Ignition Source Sends a Spherically Expanding Flame Front Propagating Into the Flammable Mixture in the Vent Stack	2 - 5
2-3.	A Propagating Flame Front Impinges on a Screen Flame Arrester Mounted on the End of the Vent Stack and Does Not Enter the Piping	2 - 6
2-4.	An Internally Mounted Screen Flame Arrester is Penetrated by an Accelerating Flame in the Vent Stack Fiping	2 - 7
2-5.	A Propagating Flame Penetrates a Damaged Screen Flame Arrester and Accelerates in the Piping	2–8
3-1.	Test Facility Fuel and Air Systems Schematic Diagram with Instrumentation Locations for Flame Arrester Testing	3 - 2
3-2.	Flashback Flame Test Chamber Flow System Schematic Diagram with Instrumentation Locations for Flame Arrester Testing	3 - 3
3-3.	Combined Air, Fuel, Vaporizer, Condenser, and Induction Systems on B-Stand	3 - 5
3-4.	Flame Sensors and Pressure Sensors Mounted on the Witness Section Piping	3 - 6
3-5.	Full-Scale Flame Test Chamber Installed on B-Stand	3-7
3 - 6.	Downstream Location of the Hydrogen/Air Spark Igniter and Flame Shield	3 - 9
3-7.	Exhaust-Burn Stack Assembly and Frangible Diaphragm at Flame Test Chamber Exit	3-1
3-8.	Sustained Burning Arrester Assembly Test Facility	3-1
4-1.	Typical Example of Transient-State Data Recorded on FM Tape and Played Back on an Oscillograph	4-6
4-2.	Hydrocarbon Gas Sample Analyser and Air Dilution Flow System Schematic Diagram	4-8
4-3.	Flame Test Chamber Motion Picture Camera Installation	4-9

4-4.	Schematic Drawing of Flame Test Chamber Meti n Picture Camera Installation	··
4-5.	Toluene/Air Mixture Flame Propagation From Idmition to Sustention on Dual 20-Mesh Screen Arrester	
4-6.	Toluene/Air Mixture Flame Propagation From Lenition to Penetration into the Open Ended Facility Pipins	
6-1.	Subscale Flame Test Chamber Installation on B-Stand	€
6 - 2.	Single 30-Mesh Screen Arrester Mounted in Fipe Spool Adapter	6
7-1.	Exploded View of Components for a Dual 20-Mesh Screen Arrester	7 - 2
7-2.	Dual 20-Mesh Screen Arrester Test Installation	7-5
7-3.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Assembly	;; <u>-</u> :-
7-4.	Crimped Ribbon Arrester Test Installation	7-4
7-5.	Packed Bed of Aluminum Ballast Rings Arrester Assembly	7-6
7 - 6.	Packed Bed of Rings Arrester Test Installation	(+)
8-1.	Screen-Type Flashback Flame Arrester Test Program Logic Diagram	8-2
8-2.	Propane/Air Mixture Using Upstream Igniter Position Test Results	8-4
8-3.	Propane/Air Mixture Using Downstream Igniter Position Test Results	5 -
8-4.	Ethylene/Air Mixture Using Downstream Igniter Position Test Results	8-6
8-5.	Ethylene/Air Mixture Using Upstream Igniter Position Test Results	8 - 7
8-6.	Gasoline/Air Mixture Test Results	8-8
8-7.	Packed Bed of Rings Arrester with Single 30-Mesh Screen and Grid Retainer Test Assembly	8-ò
8-8.	Methanol/Air Mixture Test Results	შ - 1
8 0	Toluene/Air Mixture Test Posults	85=3

8-10.	Diethyl Ether/Air Mixture Test Results	ਰ–12
8-11.	Butane/Air Mixture Test Results	6-15
€-12.	Acetaldehyde/Air Mixture Test Results	8-1-
9-1.	Typical Thermocouple Instrumentation Installation for Sustained Burning Tests	9-2
9-2.	Screen-Type Arrester Test Assembly, 15.2-cm Diameter, Schematic Drawing	9 - 3
9-3.	Single 30-Mesh Screen Arrester, 15.2-cm Diameter, Propane/Air Mixture Sustained Burning Test Results	9-i+
9-4.	Dual 20-Mesh Screen Arrester, 15.2-cm Diameter, Propane/Air Mixture Sustained Burning Test Results	9-4
9-5.	Screen-Type Arrester Test Assembly, 25.4-cm Diameter, Schematic Drawing	9-5
9-6.	Single 30-Mesh Screen Arrester, 25.4-cm Diameter, Propane/Air Mixture Sustained Burning Test Results	
9-7.	Single 30-Mesh Screen Arrester, 25.4-cm Diameter, Posttest	9-7
9-8.	Dual 20-Mesh Screen Arrester, 25.4-cm Diameter, Propane/Air Mixture Sustained Burning Test Results	9-7
9-9.	Dual 20-Mesh Screen Arrester, 25.4-cm Diameter, Posttest	9-8
9-10.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Test Assembly Schematic Drawing	9-9
9-11.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Propane/Air Mixture Sustained Burning Test Results	9-10
9-12.	Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Downstream End Posttest	9-11
9-13.	Packed Bed of Aluminum Ballast Rings with Single 30-Mesh Screen Arrester Test Assembly Schematic Drawing	9-10
9-14.	Packed Bed of Aluminum Ballast Rings with Single 30-Mesh Screen Arrester Propane/Air Mixture Sustained Burning Test Results	9-1 <

	9-15.	Packed Bed of Ballast Kings with Single 30-Mesh Screen Arrester Fosttest	9-1-
	9-16.	Spiral-Wound, Crimped Stainless-Steel hibben Arrester Ethylene/Air Mixture Sustainer Burning First Test Results	9-13
	9-17.	Spiral-Wound, Crimped Stainless-Steel Rinter. Arrester Ethylene/Air Mixture Sustainer Burning Second Test Results	91
	9-18.	Packed Bed of Ballast Rings with Wingle Ro-Mosh. Screen Arrester Ethylene/Air Sustainer Burning Test Results	()= (
	9-19.	Single 30-Mesh Screen Retainer from the Eacke: Bed of Ballast Rings Arrester Posttest	9
Table:	<u>5</u>		
	1-1.	Tabular Summary of Flashback Flame Speed and Test Chamber Peak Pressure Rise	<u>.</u>
	1-2.	Tabular Summary of Flashback Flame Quenching Test Results	1-5
	1-3.	Tabular Summary of Sustained Burning Test Results	1-6
	2-1.	Properties of Selected Fuels	2-9
	2-2.	Combustion Properties of Selected Test Fuels	2-10
	3-1.	Symbols and Descriptions for Flow System Schematic Diagram	3-4
	14-1.	Instrumentation and Calculated Test Parameter Nomenclature	·
	4-2.	Maximum Uncertainty for Measured and Calculated Parameters at the Standard Test Condition	<u>1</u>

SECTION I

CUMMARY

An experimental program was conducted to determine the flame quenching capability of four types of flame arresters suitable for installation on fuel cargo tank vents. The four types of flame arresters included a single 30-mean. screen arrester, a dual 20-mesh screen arrester, a spiral-wound, crimped ritt to arrester, and a packed bed of rings arrester. The tests simulated the exhauter flammable fiel/air mixtures from a cargo tank vent into an open deck environment. ignition of the exhaust from an external source caused a flame to flash back ver a finite run-up distance to the vent stack, which was protected by a flame arrecter. In some tests, the flame was sustained on the arrester for durations up to 30 minutes. The flashback flame tests used eight different fuel/air mixtures to produce flames with speeds representative of those from fuels that could be carried as bulk cargo aboard typical transport vessels. The fuels used in testing were (1) acetaldehyde, (2) butane, (3) diethyl ether, (4) ethylene, (5) gas dine, (6) methanol, (7) propane, and (8) toluene. Of these fuels, propane and ethylene were used during the facility check-out, the initial screening tests, and the sustained burning tests. The standard test condition was a fuel/air mixture at an equivalence ratio from 1.0 to 1.2 (which produced the theoretical maximum flame speed for the fuel used) and a flow velocity that was low enough, 1.52 m/s (5 ft/s), to assure flame propagation back into the inlet piping in the event of an arrester failure.

The experimental program was performed at the Jet Propulsion Lab ratory's Edwards Test Station (JPL-ETS) where the existing B-Stand facility provided suitable safety protection and support activities. A photograph of this test facility is shown in Figure 1-1. The facility was modified by adding a gaseous fuel system, a large flame test chamber, and a vertically directed, sustained burning test stand. The fuel/air supply and induction system provided a continuous flow of flammable mixture into a 23.8-m (78-ft.) length of 15.2-cm- (6-in.-) diameter piping. The flame arrester test assemblies were mounted at the end of the facility piping t simulate the vent stack configuration aboard a tank vessel. Optical flame sensors, pressure sensors, and thermocouples were installed in the facility piping to witness and record any flame penetration. A 2.44-m- (8-ft.-) diameter by 4.27-m- (14-ft.-) long cylindrical chamber provided a protecting enclosure surrounding the test arrester and the flow area for a considerable distance i wnstream. The open ends of the test chamber were covered with a thin oraque plactic film to prevent wind dilution and dispersion of the flammable fuel/air mixture plume, but offered minimal restriction to the expanding gases after combustion. An exhaust collector and burn-off stack located at the downstream end of the test chamber maintained atmospheric pressure within the chamber before ignition, and provided a means of reducing atmospheric pollution from the unburned fuel/air mixtures passing through the chamber. Optical flame sensors, pressure sens ro, and a high-speed motion picture camera were used in the flame test chamber t witness and record ignition and flame proparation. It was possible to ignite the the lair mixture from two different locations: (1) at the upstream end of the chamber, close to the face of the test arrester, and (2) at the downstream end of the chamber where the distance was sufficient to insure that the flame propagating upstream had achieved steady-state speed upon reaching the test arrester. The



An initial series of screening tests were made in the full-scale Clame that chamber using propane/air mixtures and ethylene/air mixtures (as representative of the two extremes of probable flame speeds for typical bulk cargo fuels): determine which igniter location (upstream or downstream) produced the most reverse test conditions. The severity being identified as the highest flame speed pr pagating upstream towards the test flame arrester. Both the single 30-mesh screen arrester and the dual 20-mesh screen arrester were evaluated for flame quenching capability on these tests. The resulting flame speeds ranged from 2.99 to 6.00 m .: (9.81 to 21.65 ft/s) with the upstream igniter location producing the higher Mame speed for both fuel/air mixtures. A tabular summary of average values of flame speeds and peak pressure rises for all fuels tested is given in Table 1-1. The single 30-mesh screen arrester quenched all flashback flames for both fuel/air mixtures. The dual 20-mesh screen arrester quenched all propane/air mixture flames and the ethylene/air mixture flames initiated by the downstream igniter location. The ethylene/air mixture flames initiated by the upstream igniter location penetrated the dual 20-mesh screen arrester in three successive test firincs. A tabular summary of the flashback flame quenching test results for all fuel/air mixtures and test arrester assemblies is given in Table 1-2.

The upstream igniter location was used on all the subsequent flashback flame quenching tests. The single 30-mesh screen arrester and the dual 20-mesh screen arrester were tested with the six remaining fuel/air mixtures. Both arresters were successful in quenching the flames on all test firings as shown in Table 1-2. The resulting flame speeds, or test condition severities, for the six additional fuel/air mixtures were less than those measured for the ethylene-fuel/air mixture, as shown in Table 1-1.

The original test configuration for the packed bed of aluminum Ballast rings arrester was unsuccessful in quenching the flashback flames from gasoline/air mixtures in three successive test firings. A single 30-mesh screen was added on the downstream end of the arrester, between the retainer grid and the bed of rings. This modified configuration was successful in quenching flashback flames from the propane/air mixture, gasoline/air mixture, and three out of four test firings with ethylene/air mixture. The spiral-wound, crimped stainless-steel ribbon arrester was successful in quenching all flashback flames from propane, ethylene, and gasoline-fuel/air mixture test firings. The test results are summarized in Table 1-2.

The sustained burning tests were conducted outside of the flame test chamber by rearranging the facility piping. Using a combination of pipe elbows, the last section of inlet pipe was redirected 90 deg to one side and the flame arrester test assemblies were mounted on the end of the pipe in the vertically up position. Two different sizes of flame screen arrester assemblies were tested, (1) the original 15.2-cm- (6-in.-) diameter adapter housing and (2) a new 25.4-cm- (10-in.-) diameter adapter housing. This change in arrester flow area was made to evaluate the effects of the approach velocity and flow-through velocity of the fuel/air mixture on the thermal environment at the screens. The single 30-mesh screen arrester and the dual 20-mesh screen arrester in both pipe sizes, the packed bed of Ballast rings arrester, and the spiral-wound, crimped ribbon arrester were all successful in maintaining sustained burning with the propane/air mixture for the full 30 minutes (1800 seconds) of test duration.

Table 1-1. Tabular Summary of Flashback Flame Speed and Test Chamber Peak Fressure Rise

Igniter Location	Average	Average Flame Speed at Tester Arrester	at Teste	r Arrester	Average 1 Rise in t	Average Peak Pressure Rise in the Flame Test
And Type of Fuel	Flame Se m/s	Flame Sensor Data, m/s (ft/s)	Photogram/s	Photographic Data, m/s (ft/s)	Cha N/m ²	Chamber, N/m^2 (psid)
Downstream	-	·				
Propane	2.99	(9.81)	3.38	(11.09)	814	(0.118)
Ethylene	4.35	(14.27)	4.02	(13.19)	931	(0.135)
Upstream						
Propane	4.75	(15.58)	3.40	. (11.15)	1043	(0.151)
Ethylene	09.9	(21.65)	4.75	(15.85)	1102	(0.160)
Gasoline	4.22	(13.85)	2,42	(46.7)	1020	(0.148)
Methanol	4.35	(14.27)	3.18	(10.43)	831	(0.120)
Toluene	5.42	(17.78)	3.21	(10.73)	899	(0.097)
Diethyl ether	5.61	(18.41)	3.73	(12.24)	937	(0.136)
Butane	3.62	(11.88)	2.90	(9.51)	936	(0.134)
Acetaldehyde	5.30	(17.39)	4.64	(15.22)	1102	(0.160)

Table 1-2. Tabular Summary of Flashback Flame Quenching Test Results

	Ĺ			Type of	f Fuel	and Nu	and Number of	Flames	s Quenched	hed						
Igniter Location And And	Propane	arie	Ethy	Et hy lene	Gasoline	line	Methano]	nol	Toluene	ane	Diethyl Ether	hy]	Butane		Acet-	de
	Yes	No	Yes	No	Yes	No	Yes	No	Yes	N _C	Yes	No	Yes	No	Yes	No
Downstream																
Dual 20-Mesh screens	٣		т													
Single 30-Mesh screen	77		٣													
Upstream																
Dual 20-Mesh screens	က			٣	m		3		m		m	-7			en	
Single 30-Mesh screen	m		٣		m		٣		<i>7</i>		m	Ψ.			(*°	
Packed bed of Ballast rings						м										
Packed bed of Ballast rings with single 30-mesh screen	œ		٣	-	m											
Spiral-wound, crimped ribbon	٣		77		11											

Sustained burning tests were also made with the ethylene/air mixture, but because of the anticipated severity of test conditions, only the packel bed of Ballast rings arrester and the spiral-wound, crimped ribbon arrester were tested. The spiral-wound, crimped ribbon arrester failed in two tests of 423 reconds and 383 seconds duration. The packed bed arrester failed on the first tests after only 43 seconds duration, and resulted in a deflagration-to-detonation transition in the arrester bed. On the second test, the packed bed arrester failed immediately after ignition due to a damaged screen. The results of the sustained burning tests are summarized in Table 1-3.

Table 1-3. Tabular Summary of Sustained Burning Test Results

Flame Arrester Type and Size	Type of Fuel	Time Duration of Burning, s	Flamethrough
15.2-cm- (6-in) diam. single 30-mesh stainless-steel sereen	Propane	1800	No
15.2-cm- (6-in) diam. dual 20-mesh stainless-steel screen	Propane	1800	No
25.4-cm- (10-in) diam. single 30-mesh stainless-steel screen	Propane	1800	Ло
2.54-cm- (10-in) diam. single 20-mesh stainless-steel screen	Propane	1800	No
30.5-cm- (12-in)diam. by 20.3-cm- (8-in) long spiral- wound, crimped stainless-steel ribbon	Propane	1800	No
	Ethylene	423	Yes
	Ethylene	383	Yes
25.4-cm- (10-in) diam. by 45.7-cm- (18-in) long packed bed of 2.54-cm- (1.0-in) size aluminum ballast ring plus a single 30-mesh stainless-steel screen	Propane	1800	No
	Ethylene	43	Yes
	Ethylene	C	Yes

SECTION II

INTRODUCTION

The U. S. Coast Guard, under the Ports and Waterways Safety Act (PL 92-30), is responsible for the safety of vessels and U. S. ports from the inherent hadari of handling petroleum products. The Coast Guard must insure that carro tanks aboard vessels are adequately protected from ignition sources that may be present on deck. Ships and barges that carry grades D and E flammable cargo are required under Subcharter D of Title 46 to have flame screens on the vent outlets of carrotanks, cofferdams and void spaces, and on all open ullage holes, hatches, in Putterwirth plates. The screens prevent accidental flame passage from the open deck into the cargo tank. A single 30-mesh screen or dual 20-mesh screens appared more than one-half inch apart and not more than one and one-half inch apart are approved by the U. S. Coast Guard.

The adequacy of the flame screen as a flame arrester has been questione: (Reference 2-1). Wilson and Crowley (References 2-2 and 2-3) carried but tests for the U.S. Coast Guard with screen arresters, where the screens were mounted some 1.83 m (6.0 ft) inboard from the open end of the pipe, rather than at the end as in the standard vent-stack installations. These nonstandard installations were used for tests of screen arresters at high turbulent flame speeds, ranging from 2 to 30 m/s (6.6 to 98.4 ft/s). These tests of screen arresters were more severe than those where the screens were mounted in the standard installation. Under certain conditions, screen arresters failed to quench the flame in some of these tests. It seems, however, that the higher flame speeds were accompanied by gross gas motions that caused apparent discrepant flame quenching results. Because the Wilson and Crowley test conditions were not representative of flashback-flame propagation to a standard vent-stack installation in an open environment, more tests that simulated the actual conditions existing aboard fuel cargo transport vessels were needed. One of the major points of interest is whether or not a flame will accelerate in an open deck environment and what effect this accelerated flame speed has on the quenching capability of the screen arrester.

Screen flame arresters mounted at the end of a vent stack are designed to prevent flames ignited outside the tank from propagating into the tank. It is assumed that the flammable gases in the vent stack are either quiescent or flowing out. On the other hand, most of the reported tests on screen flame arresters confine the flame in an enclosure whose only or major outlet was through the flame arrester (Reference 2-4). Combustion within an enclosure is invariably accompanied by considerable gas flow through the screen in the direction of flame propagation. The hypothesis to be tested was whether an unconfined turbulent flame flashback can be stopped from propagating into a vent stack whose end is covered with a screen flame arrester. In these tests, it was supposed that there is no gross gas flow through the screen associated with the ignition and propagation of the flame.

Screen flame arresters are designed to completely enclose the outlet openings with a fine wire mesh. The wire mesh is sufficiently open so that it offers negligible obstruction to the passage of gases and vapors, but the mesh openings are too small to allow the passage of flames. There should be no opening in the

screen flame arrester with an equivalent hydraulic diameter larger than the critical diameter of flame quenching in a tube. The critical diameters for flame quenching in a tube for a large variety of different flammable gas mixtures have been established in extensive laboratory tests, as discussed in Wilson and Attalah's review of flame arresters for cargo venting systems (Reference 2-5). It has been shown for laminar flames propagating in flammable gases that the correlation for the critical Peclet number (Pe) (Reference 2-h) is:

$$\log_{10} Pe = 1.8 \pm 0.3$$

Pe is defined as $D_{CR} \times Su/\alpha$, where D_{CR} is the critical diameter for flame quenching in a tube, Su is the laminar flame velocity in the unburned mixture, and α is the thermal diffusivity in the unburned mixture. The uncertainty in the value of log_{10} Pe allows for differences in the behavior of widely different fuels and exidizers, but it is sufficiently restrictive to yield useful design values for the maximum allowable opening sizes in flame arresters.

The concept of quenching a laminar flame in a narrow tube through heat less to the walls of the tube is well established (Reference 2-5). For effective flame quenching, the surface must be noncatalytic (this requirement is satisfied by all commercial materials of construction) and heat dissipative (stainless steels have adequate conductivity). Screen flame arresters differ from isolated orifices of the flame quench theory in that there are arrays of orifices. Each orifice in the array acts identically to an isolated orifice as far as flame quenching is concerned. Gas flows and heat transfer associated with flame propagation and gas volume expansion seem to be the main causes of screen failure. The flame heats and weakens the wires of the screen so that fluid friction and pressure tear openings into the wire mesh (References 2-6, 2-7, and 2-8). It is evident that prolonged exposure to sustained burning will decrease the quenching capability of the screen arrester, a phenomena that requires further investigation.

Flames propagating in open environment are almost invariably turbulent, as opposed to the laminar flames considered in the quenching theory (Reference 2-9). For most practical considerations, open turbulent flames can be considered highly wrinkled laminar flames whose characteristic wrinkle dimension is in the order of the critical diameter for flame quenching. The heat release rate is proportional to the total area of the propagating wrinkled flame front, which can be many times larger than the superficial projected flow area. The criterion for the critical diameter for flame quenching by the flame arrester is the same for turbulent and laminar flames according to Reference 2-4, but the heating effects of the turbulent flame are very much greater. In addition, the nonuniform and fluctuating turbulent flame front can cause, in pockets of the flame, the release of transient high pressure and high heat that far exceed in value the pressure and heat of a laminar flame (Reference 2-11). If a transient high reactivity pocket of gas coincides with the intersection of the flame front and the screen flame arrester, there is a probability that the flame will penetrate the screen

Equivalent hydraulic diameter = $\frac{4 \times (cross-sectional area of passageway)}{perimeter of passageway}$

at that I duti n. To prevent such flame penetration, concernative is its practice would call for screen genines substantially smaller than the the retiral critical diameter for flame quenching. Those considerations are probably the reason that Rozlevskii and Makaznov's review (Reference 2-b) presents such a wide range of critical lesiet numbers reported by different investigators in simulation of practical fire environments.

It is important to make a distinction between "burning velocity" and "plane speed" (Reference 1-9). Burning velocity is defined as the speed of the proparation of a flame front relative to the speed of the unburned gas. It is a property of the mas composition and of the physical state of the unburned gas mixture. Flame speed is defined as burning velocity plus any gross motion in the unburned mas relative to a fixed frame of reference. It is influenced by gross gas motion and by the recometry of any enclosing structure.

The proparation of a flame in a duct can create gross gas motion. This is clearly illustrated if we consider a duct, closed at one end and open to the atmosphere at the other, filled with a flammable gas. When the gas is ignited at the closed end of the duct, the flame speed is greater than it would be if the flame were started at the open end and allowed to travel toward the closed end. In the case of closed end ignition, the burned gas is expanding and pushing the unburned gas out the open end of the duct, so that the "flame speed" is the sum of the "burning velocity" and the gross motion, which is caused by the expansion of the trapped hot combustion products. In the second case, the ignition at the open end causes the unburned gas to remain stationary, hence the observed "flame speed" is nearly the "burning velocity" with differences due mainly to flame front interaction with the duct wall.

While gross gas motion does not change burning velocity by itself, there are additional factors that cause enclosed turbulent flames to accelerate in burning velocity. Acceleration of turbulent flames in ducts has been discussed in a previous JPL report (Reference 2-10) in connection with transition from deflagration to detonation. Little understood interactions between turbulent flame propagation and the turbulent boundary layer on a duct wall can lead to appreciable acceleration of the burning velocity. The flame can be accelerated to such a high speed that shock waves become associated with the highly turbulent flame front, whereupon compressive heating causes still greater acceleration until detonation is obtained. In a confined duct, particularly in those with rough walls, turbulent flames can readily accelerate to the point where self-compressive ignition occurs. The transition from deflagration to detonation in hydrocarbon-fuel/air mixtures is an extremely improbable event in an open environment, but detonations can be initiated by a shock wave from an external source, such as a bomb (Reference 2-11).

Pipes carrying vapors out of cargo tanks that contain volatile flammable liquids may contain a fuel/air mixture within the flammable range, as illustrated in Figure 2-1. A source of flame ignition outside the vent stack, as illustrated in Figure 2-2, may cause a flame to propagate into the vent stack. Flame propagation within a narrow pipe is particularly dangerous, because both confinement of the expanding hot combustion products and flame front acceleration due to interaction with the wall boundary layer can occur. In severe cases, the flame propagation can become a destructive detonation wave. The illustration in Figure 2-2 shows a flame front accelerating inside a pipe in contrast to the uniform rate. I propagation in the open air.



Titure C-1. A Flazmable Fuel/Air Mixture Flowing Slowly Duto fite Vent Stack Into the Open Air

the least stack to finded while still allowing free flow of vent vapors is shown. In single 1-1. Here the flame from an outside ignition source does not proparate for the vent stack, but impinion on the surface of the screen. If the equivalent product is clameter of the periods in the wire mesh of the screen flame arrester sections for the militar ilameter of reflame quenching, then the flame will so it product by a screen. The embeddant theory of the critical flame quench therefore has been reviewed by Wilson and Attalah (Reference 2-5). The flame, see, a wever, a stingle to burn in the surface of the screen if there is a flow of formatic mixture through it. The continued heating can lead to flame flasheds if the wire screen's temperature becomes high enough. One element of the corner was to test discoprability to flashback due to a stingle burning on the leafure of the green flame arrester.

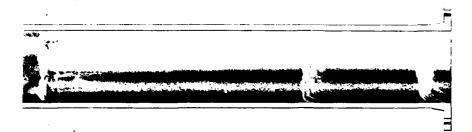


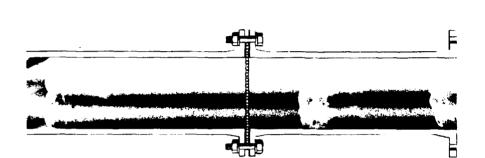
Figure 2-2. An External Tenition Course Sends a Opherically Expansion Frame Front Propagating Into the Flammable Mixture in the Vent Stack

If the pareen is mosted internally in the vent stack as shown in Fireward, the flame arresting effectiveness of the screen is requel. Up nontry into the spen ent of the vent stack, the flame accelerates. The accelerates flame specimises the pressure to rise sheaf of the flame front, and, if severe en was, push the flame through the screen. The sas flow in the pipe can be mountarily reversed a that the flame front and the sas flow is then provide in the same direction, which is the screen flame arrester. The the flame has penetrated the screen, the situation is even more named as since the failed screen flame arrester now acts as a branier to hat sas flow, and thus cruses an even greater acceleration of flame speciand a greater sikelihold of set native sumbation.



There = -. To researther Flore Front Impires on a Coreon Forme work for Module in the Eric of the Vent Stack and out Dot outer the Higher

The office are the term in the second in a hole may fail to arrest a properation of the fitted fill the fill and the fill the





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Minute (29. A fr promiting Flame Lenotrates a Damaged Jureen Flame Appeated and Appelerates in the Hiping

explicitude the circular of at at indimetric air fue, ratio, and the second is at the ribble of the partonial dimeter—equivalence rationary. For all practical correct, the equivalence ration of maining a penching diameter coincides with the solitation result of a maximum condition to edity. The spontaneous immitten temperations are respective in an ACTI test is given. The equivalence ration for the lean edge of the left of a gweet proparation in a closed tube is listed. These values are well to describe the conditions of a familiar in the flame arrester tests, and the action in the graph of a condition of the pentagon.

Table 2-1. Properties of Selected Fuels

Common Name	Chemical Name	Formula	Molecular Weight
Acetaldehyde	Ethanal	снзсно	44.053
Butane	n-Butane	C4H10	58.123
Diethyl ether	Ethoxy ethane	(c ₂ H ₅) ₂ o	74.122
Ethylene	Ethene	^C 2 ^H 4	28.054
Gasoline	_	^C 8 ^H 15.44	111.44
Methyl alcohol	Methanol	сн3он	32.042
Propane	Propane	^C 3 ^H 8	44.096
Toluene	Methyl benzene	^C 6 ^H 5 ^{CH} 3	92.140

Table 2-2. Combustion Properties of Selected Test Fuels

The second secon

							1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
	Stoichiometric (\$\phi = 1.0\$)		Equivalence Ratio = (\phi) at	Quenching Diameter of Tube, cm	meter cm	Spontaneous Ignition	Lean rismmablilty Limit for Upward Propagation in
Fuel	Mass Ratio	cm/s	Velocity	Stoichiometric Minimum	Minimum	oc (oF)	φ φ φ
Acetaldehyde	7.85	(50)°	(1.15) ^c	0.35	P	175 (347)	0.50
Butane	15.46	511	1.13	94.0	0.28	430 (807)	0.54
Diethyl ether	11.19	Lt _t	1.15	0.38	0.31	186 (366)	0.55
Ethylene	14.79	81	1.15	0.20		(116) 061	0.41
Gasoline	14.62	40 to 42	1.10	!		371 (700)	09.0
Methyl alcohol	6.47	26	1.01	0.28	0.23	470 (878)	84.0
Propane	15.68	94	1.14	0.31	0.28	204 (940)	0.51
Toluene	13.50	41	1.05			468 (1054)	0.43

 $^{^{8}}$ Composition of air: N₂, 78.087; 0₂, 20.946; C0₂, 0.033; Ar, 0.934, in volume percent.

bat one atmosphere, 25°C mixture.

Estimates.

d Data not available.

SECTION III

TEST FACILITY DESCRIPTION

A. GENERAL

All testing for this program was performed at the B-Stand facility of the Jet Propulsion Laboratory's Edwards Test Station. The B-Stand test area contains an air compressor system, fuel system, fuel vaporizer and condenser loop, fuel and air induction system, facility piping, test flame chamber, and an exhaustburn stack. The test facility flow system schematic diagrams are shown in Figures 3-1 and 3-2. Table 3-1 gives a description of the symbols used in the schematic diagrams. A detailed description of the major portion of this test facility is given in Reference 2-10. Some modifications and additions were made to incorporate gaseous-type fuels, flashback flame testing, and sustained burning testing for this program.

The following is a brief description of the various facility systems indications indications.

B. AIR COMPRESSOR SYSTEM

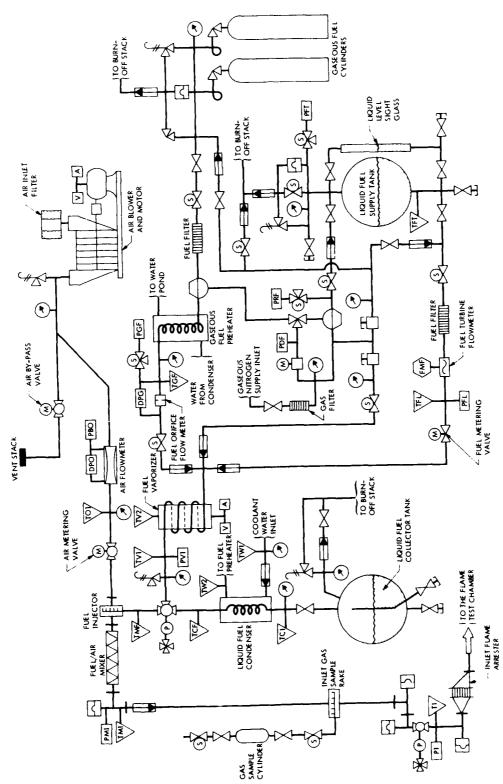
A new multistage centrifugal turbine air compressor was installed, which is rated for 11.3 m³/min (400 indicated cfm) at 41.4 kN/m² (6.0 psid). It is driven by a 14.9-kW (20-hp) electrical motor. Air flow in the 10.2-cm- (4-in.-) diameter pipe system is controlled by a remotely operated metering valve and a remotely operated bypass valve. Flow rate is measured using a Meriam Laminar Flow Element (IFE).

C. FUEL SYSTEM

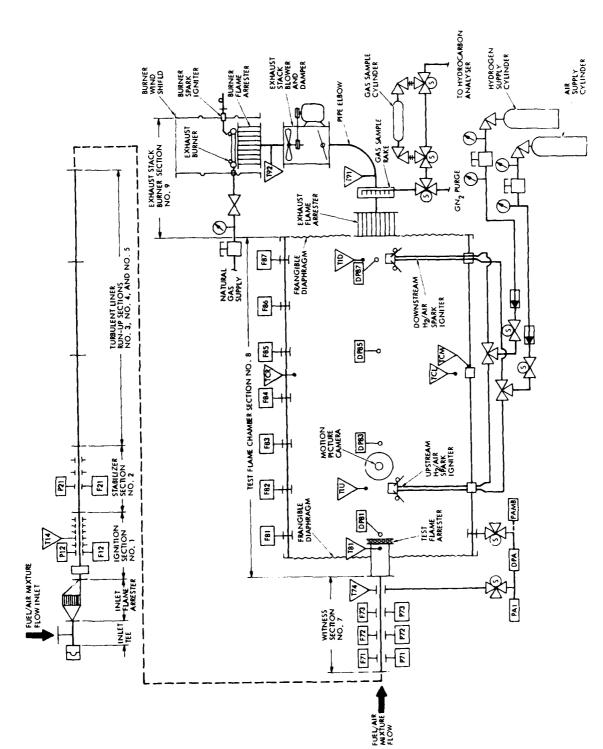
Two parallel systems provide a variety of either liquid or gaseous fuels. Liquid fuel was supplied by a nitrogen gas pressurized tank with a capacity of 0.049 m³ (13 gal) and a working pressure of 6895 kN/m² (1000 psia). Fuel flow was controlled by a remotely operated metering valve and measured with a turbine-type flowmeter. Gaseous fuel was supplied from a manifold containing two type-lA shipping cylinders having the combined volume of 0.0876 m³ (3.08 ft³). The normal delivery pressure was 8274 kN/m² (1200 psia). Gas flow was controlled by a remotely operated pressure regulator and measured with a precision-bored sonic orifice. The fuel gas temperature was stabilized for flow measurement using a water bath preheater.

D. FUEL VAPORIZER AND CONDENSER LOOP

All fuels were either vaporized or preheated with a remotely regulated electrical heat exchanger before injection into the flowing air stream. A pneumatically operated three-way valve energized to the RUN position directed the heated fuel into the fuel injection manifold. With the valve in the CONDENSER position, the heated fuel was directed into a water bath heat exchanger where most of the



Test Facility Fuel and Air Systems Schematic Diarram, with Instrumentation Locations for Fiame Arrester Testing Figure 3-1.



Flashback Flame Test Chamber Flow System Schematic Diagram with Instrumentation Locations for Flame Arrestor Testing Figure 3-2.

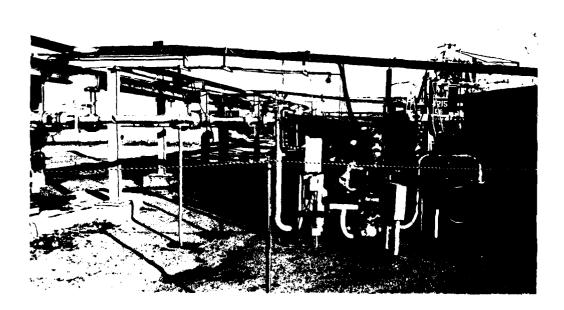
Table 34.. Tymb 4s and lescriptions for Flow Tystem Schematic bia-gram

Jymo (a	Description -
\bowtie	Manual globe valve
$ \mathfrak{S} $	Electric solenoid operated valve
	Electric motor operated valve
@ 	Electric motor operated ball valve
№ -@-∑	Air piston operated ball valve
	One-way flow check valve
₹	Pressure relief safety valve
\ominus	Dome pressure regulator valve
台	Manual set pressure regulator valve
	Electric motor operated pressure regulator valve (dome loader)
	Pressure rupture disc assembly
②	Fressure gage
∇	Voltmeter transducer
A	Ammeter transducer
4	Temperature transducer
PXX	Pressure transducer
FXX	Flame sensor transducer
FXX	Flowmeter transducer

The miner of a conservation of the respect to the state of the state x_i and x_i

Fig. was injected in the air street the up of desent as explict in the No. -one wellness there is in in the No. -one wellness the object of all into the order of the well-explicitly entire three flux one in the well well-explicitly entire the well we call the well-explicitly entire the well-explicitly and the weak precision of the velocity of the explicit the flux of the control
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the factories of this piping is directly the test section according to the feature-order to the piping is directly in a section of this piping is directly in a section of this piping is directly and instruction of this piping is directly and instruction of this piping to the registry piping is described. The injection of this piping is described and the interpretation of the section of the piping is described as a directly and the property of the section of the piper section. There is a dispersion of the section in the injection of the piper section of the section of the piper section of the personnel of the personnel of the piper of the section of the personnel of the piper of the pip



Florate (=1. 10mm line) Mary Flora, Vag miden, Commenter, majorate in Apprendiction = Attended



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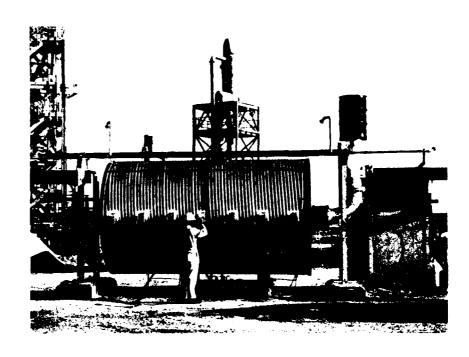


Figure 7-3. Figure 1 and least firefred courts to a make the state of the firefred courts as the state of the

0.15 mm (0.006 in.) thick and bander into place, were used as francisce dispersion to close the per shancer ends. This serves two purposes: the mark environment enhanced motion picture photography of the Clame for ht, and the subsections eliminates dispersion of the foot air platture that situations have been named by 1 ms. Winds. Once the mixture in the summore was limited, the heat and increases prescure from the burning mixture low out the displacement.

The interior statement of the summaner was painted that itsek to ship it toraging. A shape viewing part mean the aproximation of the summaner was used to take motion pictures of flame propagation. Reference which parts in the wasterp site the common were used to indicate distances as not the flame parts. Instrumental in in the test shamber includes that pressure send on a sate, equippered, in the wasterpoint the horizontal center line. We went flame send to were reliablely to also, equippeded, along the horizontal sender line or write the pressure sense of we ever, early in the prospect, the flame send or were recorded to the tip senter sine of the immediate of a better viewing position of the stratifies flame if not become used to measure the tensor as are shown in the charges. It scatters of the pressure, temperature, and there went no are shown in the scatters.

H. THE BUILDING THARD IN MITTER

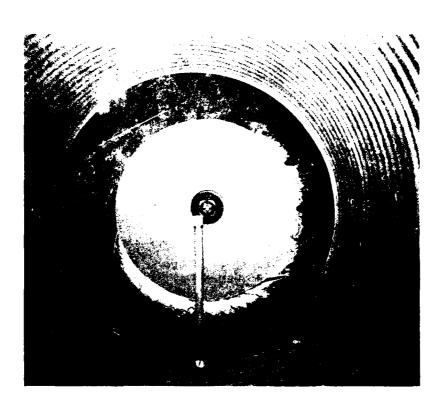
a mitter of the fact air mixture in the test chamber was accountable; with a grandentair operk isniter. This isniter recembled a small rocket obside. where interpeding jets of hyproren was and air were ignited by a spack just in the base of the embustion chamber. The resulting flame was directed sorting by where the win a short nozzle for a nominal duration of 200 mm. The imiter respects less were smilt into the end of a i.l-m- (3.5-ft.-) | not section of 5.5r-cm-'-in.-' dismeter pipe mounted into fittings on the bottom of the test chameer. he point I watten of the ignition flame was just below the axial centerline of the magner. There were three possible locations for the igniters: (1) upstream here the test arrester, (2) midchamber, and (3) downstream at the chamber exit. may the opatream and downstream igniter position were used during the test in -Fram. When the downstream igniter position was used, the francible disphratm in the chamber exit was shielded from the flame by a sheet of aluminum ecvering where ximately 1.0% of the total exit area. The aluminum shield delayed the runture I this diaphragm until the flame had traversed the length of the chamber t reach the test arrester on the inlet end. This delay made it possible to their I is madity motion pictures of the flame impinging on the arrester before the mander was exposed to ambient light through the ruptured diaphragms. A ph t with it the downstream igniter and flame shield are shown in Figure 3-6.

EAHAUNT-BURN STACK

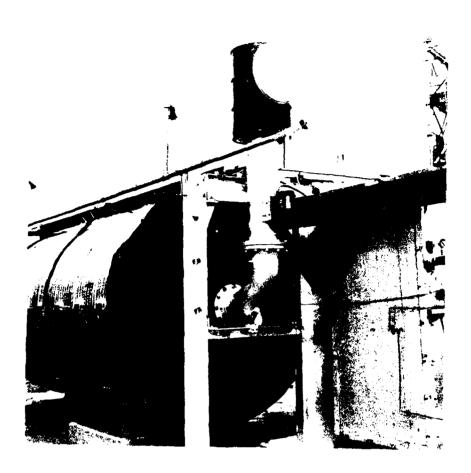
and exhaust-burn stack was required for this test facility in compliance with simple at its more resulations covering the controlled release of hydrocarbon vapors. Into was assemblished by installing a 1.22-m (4-ft.) length of 30.5-cm (12-in.) diameter piping on a vertically directed pipe elbow at the exit end of the flame counter. The pipe contained a ducted fan and damper valve to control the exhaust of w. which, in turn, maintained atmospheric pressure in the test chamber prior to limit in. Opinal-wound, crimped metal ribbon arresters were attached to to the each of the exhaust stack assembly to prevent the propagation of flame into the eights. A crass sample rake was installed just downstream of the inlet flame expected. The fuel/air mixture sample taken at this location was fed into m. notice to take hydrocarbon analyser. The sample line was closed by a sclene is perated valve just prior to ignition to protect the analyser. At the top of the vertical stack, a shielded natural gas fired burner disposed of all combust-free expects of facts. A photograph of the exhaust-burn stack assembly and the formula of lightnarm at the exit of the flame chamber is shown in Figure 8-7.

V. PRITAINED BURNING TEST FACILITY

The facility piping was modified after completion of the flame chamber testing the release the flame arrester test assembly out to an open area for the rectained turning tests. Two pipe elb ws were inserted just upstream of the witness
section to 1 wer and turn the piping 90 deg away from the supporting structure.
An open line elb w was inserted between the downstream end of the witness section
to the arrester test assembly; this elbow directed the exhaust flow vertically
appearance, who in Figure 3-8. The gas sample rake for the hydrocarbon analyses was
the proceed to ween the flames upstream of the test section. Ignition was



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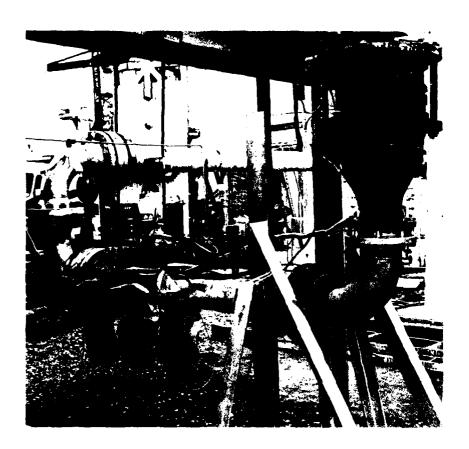


Figure 4-2. North-likes man in reservoir encourse to μ . Described a like

SECTION IV

INSTRUMENTATION AND CONTROLS

A. GENERAL DESCRIPTION

All instrumentation and controls at B-Stand facility were remate papers of and monitored. Test system parameters were measured at the test site with their signals conducted to the blockhouse for an initialization, recording, and display. Location and identification of all principal instrumentation parameters and controls are shown in Figures 3-1 and 3-2. Takes well is a listing of the nomenclature for all instrumentation and calculated parameters.

Test system parameters were divided into two groups: (1) steady-state (wespeed), and (2) transient-state (high-speed) data. Steady-state data increase all the measured and calculated parameters for the air system, fuel system, fuel system, fuel system, and the pre- and posttest pressure loss measured across the test arrester. Transient-state data includes the measured and calculated flame speem and jeak pressures developed in the test flame chamber and facility piping, and the purcess or failure of the experimental flame arrester.

Steady-state data was recorded and calculated on the JPE-develope. Integrate: Digital Acquisition and Controls System (IDAC) with back-up by the new Edwards Digital Acquisition and Control System (EDAC). Transient-state data was recorded on two high-frequency FM tape recorders and played back on an oscill smaph at an expanded time scale. Flame speeds and peak pressures were manually scaled and calculated from the oscillograph traces. Flame speeds in the test chamber were also estimated from the high-speed motion picture films.

All critical control functions were either manually positioned on the controls console or automatically operated by the preset sequence timer. These operations were selectively recorded using electrical contact closures on ILAC, EDAC, FM tape, or a second high-speed oscillograph. Two strategically place: television (TV) cameras, with video displays in the blockhouse, monit red the fuels system area and the test flame chamber. Two high-speed motion picture cameras also recorded events both inside and outside the test flame chamber contact the actual test firings. Visual coverage and controlled access to the test area were maintained by a safety monitor in an observation tower located over the blockhouse.

A detailed description of the instrumentation and centrols system is given in Reference 2-10. Modifications and new additions that were made to the systems for this test program are described in the following paragraphs.

B. STEADY-STATE DATA

The EDAC system is a new digital instrument recently installed at FTT. It was still in the process of functional checkout at the time of this program, so it was used as a backup steady-state data computing and recording system for the

where x = x + 1 is a terminant of \hat{x} in remark to include the anticonfiguration of \hat{x} and \hat{x}

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1 1r	all mo (psig)	ladecus fuel presmre		
111	wii mi (psi i)	Jaseous fuel differential pressure		
	1.1 (1)	Gaseous fuel temperature		
	kii m ^o (psig)	Fuel vaporizer outlet pressure		
3	90 (SF)	Fuel vaporizer outlet temperature		
	$\sim (oh)$	Fuel vaporizer core temperature		
4.00	a. Achl	Fuel injector inlet temperature		
* · · · ·	ku m ² (psig)	Fuel/air mixer outlet pressure		
er er er i	°C (°F)	Fuel/mixer outlet temporature		
and the second	îd (9F)	Fuel condenser inlet tempera ure		
er tra	nA (°#)	Fuel condenser outlet temperature		
	ာ () မြော်)	Coclant water inlet temperature		
277	ခြင်း (မြော့်)	Coolant water outlet temperature		
	kl./m (psig)	Inlet tee pressure		
	01 (PF)	Inlet tee temperature		
1	ki mê (paig)	Inlet section pressure		
P/I	kilm (psig)	Stabilizer section pressure		
101	kli/m² (pvir)	Witness section inlet pressure		
	kN/m/ (psig)	Witness section center pressure		
F7t	kli,m' (psig)	Witness section exit pressure		
: -	2 (sec)	Inlet section flame sensor		
F. 1	s (sec)	Stabilizer section flame sensor		
		Witness section inlet flame sensor		
#15.2 #15.2	s (sec)			
All self Efficiency	s (sec)	Witness section center flame sensor Witness section exit flame sensor		
	r (sec)			
195. 182.	kli/m² (psi i)	Flame chamber differential pressure, Sta. 1 Flame chamber differential pressure, Sta. 3		
* "	kmin(noid)	· · · · · · · · · · · · · · · · · · ·		
* 1 m. ** - 7 /2 * - 7 /2	klidmi (psid)	Flame chamber differential pressure, Sta. 5		
12:57	kH/m ^d (psid)	Flame chamber differential pressure, Sta. 7		
Prof.	s (sec)	Flame chamber flame sensor, Sta. 1		
	s (sec)	Flume chamber flame sensor, Sta. 2		
H. 4.	u (see)	Flame chamber flame sensor, Sta. 3		
Po.	ပ (ဗမင)	Flame chamber flame sensor, Sta. 4		
### ₁	a (che)	Flame chamber flame sensor, Sta. 5		
1. d.	$s_{-}(se_{2})$	Flame chamber flame sensor, Sta. 6		
#AT	a (ren)	Flame chamber Clame sensor, Sta. 7		

Table 4-1. Instrumentation and Calculated Test Parameter Nomencleture (Continuation 1)

:

Steady-State Parameters	Units, S.I. (Engr.)	Description		
T74	°C (°F)	Witness section exit temperature		
T81	°C (°F)	Test arrester inlet temperature		
TIU	°C (°F)	Upstream igniter flame temperature		
TID	°C (°F)	Downstream igniter flame temperature		
TCR	°C (°F)	Flame chamber roof temperature		
TCL	°C (°F)	Flame chamber lower temperature		
T'CM	°C (°F)	Flame chamber metal temperature		
T91	°C (°F)	Exhaust stack inlet temperature		
T92	°C (°F)	Exhaust stack exit temperature		
HCA	%	Exhaust stack total hydrocarbon analysic		
PAl	kN/m_0^2 (psig)	Test arrester inlet pressure		
DPAl	kN/m_{O}^{2} (psid)	Test arrester differential pressure-pretest		
DPA2	kN/m ² (psid) kN/m ² (psia)	Test arrester differential pressure-posttest		
PAMB	kN/m (psia)	Test area ambient pressure		
Calculated Parameters	Units, S.I. (Engr.)	Description		
MA	kg/h (1b/h)	Air mass flow		
MF	kg/h (1b/h)	Liquid fuel mass flow		
A/F	ratio	Air mass flow to liquid fuel mass flow ratio		
MFG	kg/h (1b/h)	Gaseous fuel mass flow		
A/FG	ratio	Air mass flow to gaseous fuel mass flow ratio		
φ΄	ratio	Equivalence ratio		
VA	m/s (ft/sec)	Air flow velocity through 15.2-cm (6.0-in) diameter pipe		
FXX-FYY	m/s (ft/sec)	Average flame speed between two adjacent flame sensors		
SX-SY	m/s (ft/sec)	Average flame speed between two adjacent light ports or a light port and the test arrester obtained from the motion pictures		

some of aparton, was as it will eventually separed. The neart of the EDAS Is a cata dense. It is not because of the rate up to 20,000 means a per second, set talk time any 200 denness are appired to the 8-2 test factors.

In the confirmal container than the name of the container of the container and then uses these contains at most with the container container and then uses the entineering units for each respective, and container also performs calculate the entineering units for each container, polynomials, or see see the countins involving two containers, parallel data, and contain the containers that the transfer to the nearest millipecani.

The distribution are used for record storage, from which posttest playtacks of layor permaneters and calculations are made to the line printer. The line printer nearly layorameters with channel identification, engineering units, and time. The climb programmable for maximum output at points of interest. The view that no provide on-line real-time displays of up to 10 parameters with channel limiting programmable for maximum output at points of interest. The view that no provide on-line real-time displays of up to 10 parameters with channel limitification, engineering units, and contact closure status. These displays we have seen to form 6 preprogrammed pages. High and low limits can be assigned in a parameters. The limit output signal is capable of operating control closure in a decimal alarms within 10 milliseconds of exceeding a limit.

Indicate as steady-state data from both the IDAC and EDAC systems were very emperated, withouth not exactly alike. This discrepancy is reportedly cause: by a real confidence in the time base for computer calculation between the two systems that manner be resolved. The EDAC system when totally operational will require IDAC as the primary data system for follow-on programs.

T. REMITTENTATE PATA

New flame send is that enall withstand repeated exposure to ambient light and still remain sensitive to the light-blue color of hydrocarbon flame had t be amoralised the flashback Clame chamber. The Du Mont Type 6291 photomultinotice this a used as flame sensors in the facility piping were not suitable remade the phospherescent mouting on the detector can be deteriorated by bright can limit. This woltain type detectors, similar to those reported in Reference . - 3 which is not have this sensitivity, were used instead. They are the EJ and I The FUV- COR vision photovoltain detectors that have a spectral range from % to 1.1 % CoO to 1150 Gm) with a maximum response at 9000 % 900 Nm' and a sequentially of 12 \times 10 $^{\circ}$ value watt. perational amplifiers were built to the to differ into and directiony convented by EG and G. The detector and amplifier were absenced in a weather-tight alregings box with a phototube viewing port subtaining a simple from the distance from the collimating The translation will be varied to primize the viewing angle and detect of Thus of son in. As how how the rise-time response of the photovoltaic detect in is a new ment as were then the photomultiplier tube, 1.5 microseconds compared to the the versus, it is a re-than adequate for setecting the expanding atmospheris Misses the state to the flame thursber.

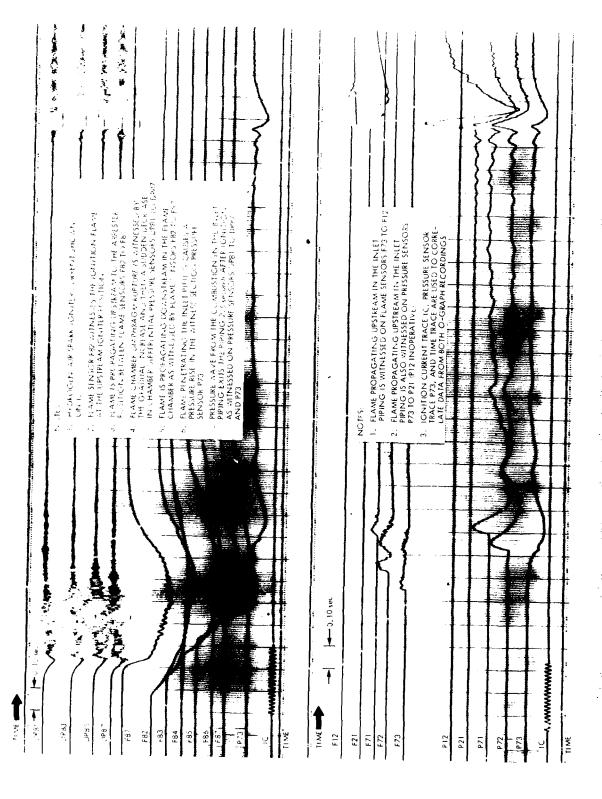
Notes phit wiltair flame consider were initially installed at the -most property intervals along the horizontal contorline of the flame species. They were stering outside to the chamber to center income much plot area of flame propagation on weather flame illumination intensity varying unprelistably from top to be to make the chamber. It is believed this in caused by gravitational stratification of the flam flame after it leaves the facility piping. The flame setect of vertical view, looking a who into the propagation flame front, resulted in more reliable flame speed measurements.

Frame chamber peak pressure rise was measured with four Statham Model HV 5 In differential pressure-type transducers mounted at 1.02-m (A-ft) intervals as not the horizontal centerline of the chamber. It was intended that these pressure density measure the pressure rise at the chamber wall during the passage of the flame from the pressure in the flame from the spherically expanding ball of flame up to the point of manner diaphrasm rupture. The resulting resonance from this pressure spike in the manner masked any evidence of flame passage past the individual pressure contains.

Intervals on apposite sides of the witness section piping were used to record classificate flame penetration through the test arresters. Two of these flame send to were a photomultiplier tube type and one was a photovoltaic type. The prescare censors were all quartz-crystal piezoelectric-type transducers flush-mounted to the inside wall. In addition, there was a similar combination of flame sensor and pressure sensor in both the inlet igniter section and stabilizer section of the facility pixing to record flame propagation up to these locations. An inlet flame arrester stopped any further flame penetration beyond this point into the invertion system piping.

The signals from all flame sensors and pressure sensors located in the flame chapter our facility piping were recorded on two high-frequency FM tape recorders and the haline scallegraph. A 100-Hz coded time pulse and the spark igniter surrent were also recorded and used as reference points for test initiation and time a record in between the various recorders. A typical example of transient-state rate if in the flame chamber sensors recorded in the FM tape and playback on an ordinarraph with an expanded time base is shown in Figure 1-1.

The ELAS system was used as the principal recorder for the thermal soak-back rata measures by thermocouples instabled in the flame arresters during sustained coming tests. See ried at millisecond soan intervals, the data was time edited, player back, and printed at time intervals ranging from 5 to 120 seconds, depending in the length of test and the transient-state of the data. Video displaye forms real-time flame arrester temperature mad were monitored during the test to identify flame penetration. This was later confirmed by data from the flame senders and pressure sensors in the witness section piping that was recorded on FM magnetic tape and played back on the action graph.



D. GAS-SAMPLE ANALYSIS SYSTEM

The das-sample analysis system used for this program is describe: in letal. in Reference 2-10. Briefly, it is an on-line system that utilizes a hocknow. When 400 Total Hydrocarbon Analyser instrument combined with a JFL designed and foldicated air dilution and calibration system. The analyser automatically and the tinuously measures the concentration of hydrocarbon in a flowing gas comple, utilizing the flame ionization method of detection. It was calibrated using prepane (C3H8) and air mixtures. To analyse other hydrocarbon fuel and air mixtures, the number of carbon atoms per molecule of fuel had to be in a ratio to that of propane. A flow system schematic drawing of the complete gas-sample analysis system is shown in Figure 4-2. A listing of the fuels and their properties that are used in this program is given in Tables 2-1 and 2-2.

The hydrocarbon gas analyser was located as close to the test flame chamber as practical to minimize response time. It was placed in a steel-walled protestive enclosure adjacent to the exhaust-burn stack. The gas sample rake was installed in the inlet elbow of the exhaust-burn stack. A three-way splenoid valve provided a gaseous nitrogen purge through the sample rake when not in use. Analyser response time after activation of the three-way sample valve was approximately 30 seconds. Figure 3-7 is a photograph of the protective enclosure housing the gas analyser located next to the exhaust-burn stack.

E. PHOTOGRAPHIC DATA

Two motion picture cameras were used to record every test firing. One camera was positioned outside the flame chamber with a view of the entire test section assembly. Operating at 32 frames per second, this camera recorded the rupture of the flame chamber diaphragms and the extent of the emitted flame plume. The other camera was positioned adjacent to the flame chamber observation window with a view of the inside of the chamber, including the upstream igniter and the downstream face of the test arrester. Figure 4-3 is a photograph of this camera installation. Operating at 100 frames per second, it was possible with this camera to record the propagating flame front inside the chamber. Four light ports, equally spaced on the opposite wall, provided reference points for determining distance traveled. A schematic drawing of the flame-chamber camera installation is shown in Figure 4-4. The distances traveled by an expanding spherical flame, when viewed by the camera, are indicated between each adjacent light port, and from the light port in line with the igniter to the face of each of the four flame arrester test assemblies. By counting the number of motion picture frames required for the flame front to traverse these known path lengths, the lapse time was estimated and the average flame speed was calculated. The flame speeds obtained by this method will not necessarily agree with those calculated from the flame sensor data, because of the different sight locations and viewing angles, but they are of the same order of magnitude. Figure 4-5 is a selected series of six photographs taken from test motion picture film showing a tablene air flame propagation from ignition to sustention on the downstream face of the dual 20-mesh screens arrester. Figure 4-6 is a similar scries of phot graphs showing a toluene/air flame propagation from ignition to penetration into the open ended facility piping, causing an eruption of flame from the pipe.

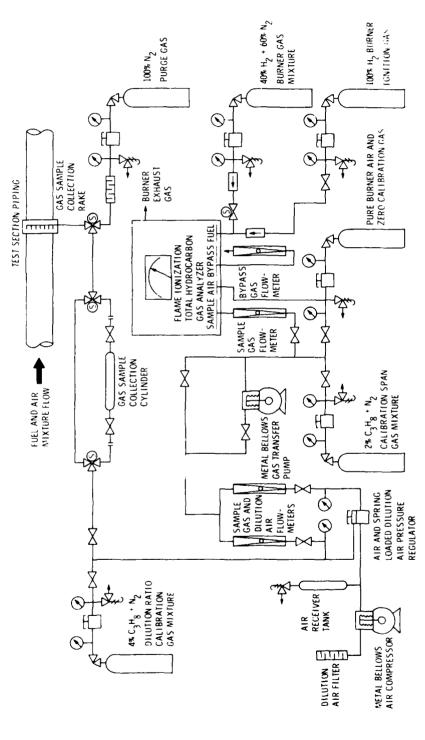


Figure 4-2. Hydrocarbon Gas Numple Analyser and Air Illati n Flow System Schematic Playram

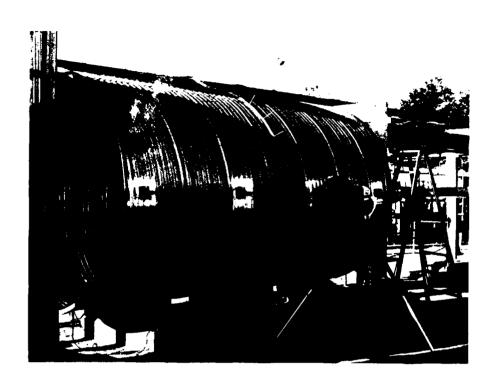
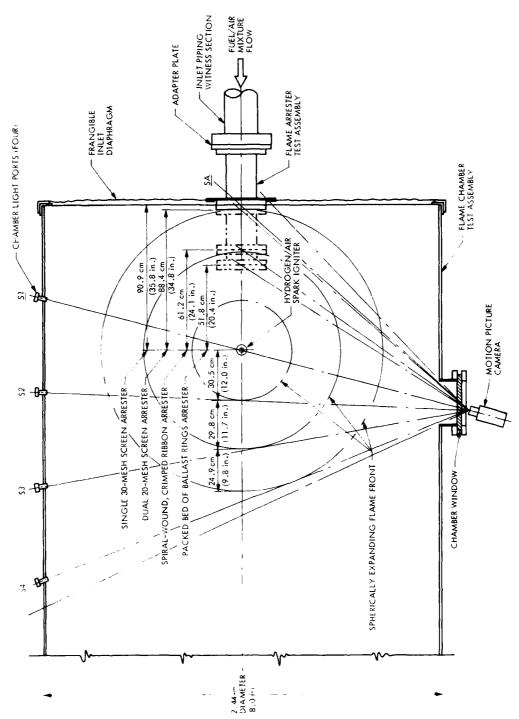


Figure 4-7. Figure Form improved this military importance to Installation

F. HARMETER MEASUREMENT OF BUILDING CONTROL CONTROL

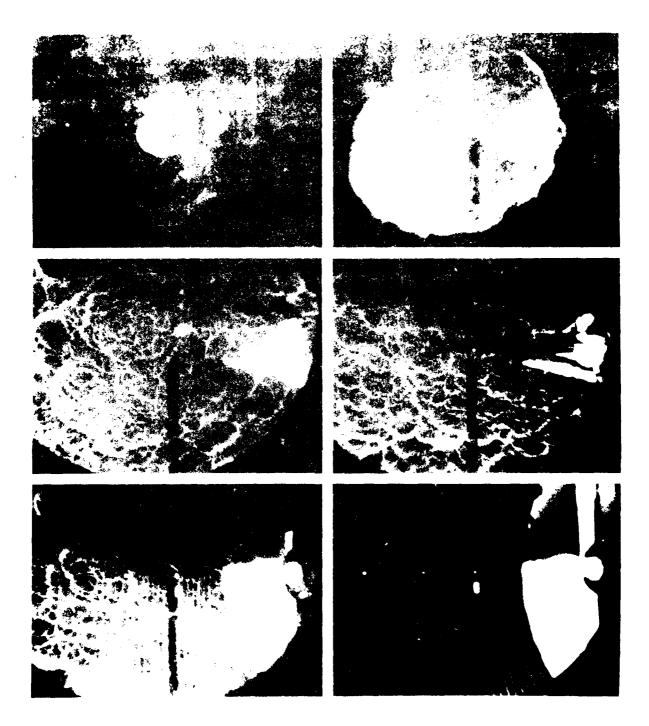
The HAT system was the originary for open of a steage-state data, with back-up by the FLAT shelper system. There systems have singular maps lifty that a night of an explosion of the printers and view ministers. Special TDAT and HAT softweether are seen were written if a air and fuel systems rate to mass succeeding and it, we have been written if we air and fuel systems rate to mass succeeding as a construction of we have been in a systems, an end-to-end randomatic nearly for a set of an establish in a systems, an end-to-end randomatic nearly and the set of a section of the instrumentation systems, randomatic nests or, and the attendance which is a measurer and randomatic parameters of presented in between end and randomatic parameters of the lab of easy-crate rate and contains are problem. For a wint is a system of the lab of easy-crate rate and contains are problem with a 1971 19 problem.

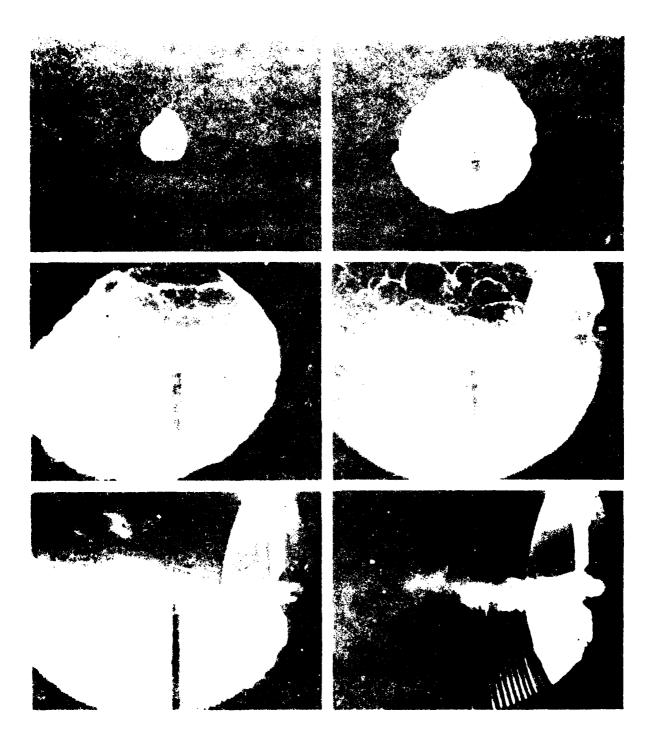
- Fig. . Undertainty of a parent are near wearest in $t \sim e^{2\pi i t}$ of tarable when the $\pm t$ rate t
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- (b) $37.8 \text{ to } 93.3^{\circ}\text{C} (100 \text{ to } 200^{\circ}\text{F}) = \pm 1.4\%$
- (c) 93.7 to 148.9°C (200 to 300°F) = $\pm 0.85\%$
- (1) $148.9 \text{ to } 204.4^{\circ}\text{C} (300 \text{ to } 400^{\circ}\text{F}) = \pm 0.65\%$
- (e) $204.4 \text{ to } 276.7^{\circ}\text{C} (400 \text{ to } 530^{\circ}\text{F}) = \pm 0.49\%$
- (f) 276.7 to 1260°C (530 to 2300°F) = \pm 0.43%
- (h) Uncertainty for air-velocty or air-mass-flow calculations in ±1.82% of value.
- (5) Uncertainty for liquid-fuel-mass-flow calculation is ±1.93% of value.
- (6) Uncertainty for gaseous-fuel-mass-flow calculation is ±2.88% of value.
- (7) Uncertainty for calculated air-to-liquid-fuel mixture ratio and equivalence ratio is ±2.65% of value.
- (8) Uncertainty for calculated air-to-gaseous-fuel mixture rationard equivalence ratio is ±3.41%.

Using the uncertainties listed above, the maximum uncertainty that can be expected for the measured and calculated steady-state test parameters associated with the average value at standard test conditions are listed in Table 4-2.

The transient-state data were recorded on an Ampex Model FR 2200 and an Ampex Model FR 3020 high-frequency FM tape recorders. Photovoltaic detector flame sensors were the primary instruments used to determine flame speeds. Strain-gauge-type differential pressure transducers were the primary instruments used to measure peak pressure rise in the flame chamber. Flame sensor and pressure sensor test data, along with pre- and posttest calibrations recorded on the FM tapes, were played back on an oscillograph at an expanded time base. The following is an analysis of the uncertainties associated with transient-state data assured with a 95% (20) probability.

- (1) The uncertainty of flame chamber peak-pressure rise measurement is ±5.85% of transducer range.
- (2) The uncertainty of calculated flame sensor flame speed measurement is ±5.45% of value.
- (3) The uncertainty of calculated photographic flame speed measurement is ±10.07% of value.

The maximum uncertainty that can be expected for measured and calculated parameters associated with the averaged values of flashback flame speed and peak pressure rise in the test flame chamber at standard test conditions are listed in Table 4-2.

Table -- . Maximum Uncertainty for Measured and Tabrulated Farameters at the Standard Test Condition

s remark toor	Symbol	Uncertainty
transgr=strate Date		
Alr flowmer or inlet pressure	PBO	±0.27 kU/m ² (+6.039 pulk
alr flowmeter differential pressure	DPO	±0.0083 kN/m ² (±0.0000 ju)
Air to wheter exit temperature	101	±1.3°0 (±2.8°F)
. Iquin fuel lime pro sure	PFL	114.0 kN/m ² (±1.0 psis-
liplicated line tomperature	TFL	±0.9°C (±2.7°F)
Hipticia, Clowmeter frequency	FMF	±0.8 Hz
Parents fuel line pressure	PGF	±17.2 kN/m ² (±0.50 psig)
Fig Pael line temperature	TGF	±0.9°C (±2.7°F
Aut arrester inlet pressure	PAl	±0.269 kN/m ² (±0.049 psig)
lest arrester differential pressure	DPA	±8.3 N/m ² (±0.0012 psia)
Terr area ambient pressure	PAMB	±0.538 kN/m ² (±0.076 psia)
Alp-mops flow	MA	$\pm 1.90 \text{ kg/h} (\pm 4.19 \text{ lb/h})$
Alm velocity	VA	±0.083 m/s (±0.27 ft/s)
limaid-fuel-mass flow	MF	±0.158 kg/h (±0.35 lb/h)
Alr to liquid-fuel-mass ratio	A/F	±0.35
Pageous-fuel-mass flow	MFG	±0.240 kg/h (±0.529 1b/h)
Air to geseous-fuel-mass ratio	A/FG	±0.44
Lagivelence ratio	φ	±0.04
leansient-State Data		
Plane chember peak pressure rise	DPXX	±121 N/m ² (±0.0176 psi.i)
Flame semsor flame speed	FXX-FYY	±0.19 m/s (±0.63 ft/s)
Histographic flame speed	SX-SY	±0.35 m/s (±1.16 ft/s)

SECTION V

TEST OPERATING PROCEDURES

A. JEMERAL CAPTUY REQUIREMENTS

All test govating procedures involving fuel transfer, or performed with the fuel system pressurized, required the safety tower operator to be in a sition, monitor all communication on a headset, and control access to the test area with the safety status lights. The test stand was normally in a Shakki condition, which permitted open access to all personnel. Fuel transfers and test preparations were performed in an AMBER condition, which restricted non-perating ferm head to the workshop area, unless permission was granted to enter them areas. A RED condition, which isolated the test stand and the surrounding designated area from all personnel, was used during actual test.

A minimum of two men was required at the site during fuel transfers and test preparations. Personnel safety equipment included hard hats, face shields, gloves, fire retardant coveralls, and for some fuels, breathing air systems. Additional safety equipment was available including safety showers, eye washes, and the Firex water deluge system. All operations, except the replacement of the flame chamber diaphragms and the changing of the test flame arrester, were performed using formal procedures in the form of check lists, with individual pages dated and timed, and with each step initialed by two persons witnessing the event.

An ignition-completion key switch, which prevented the actuation of the hydrogen/air spark igniter except during checkouts and test operations, was located at the test stand.

B. OPERATING PROCEDURE CHECK LISTS

The following is a description of the operating procedures and check lists used in the flashback flame tests.

Pretest System Checkouts

- a. Ireliminary Check. This check confirmed proper installation of the test item, instrumentation and control cable connections, readiness of the nitrosen pressurant and purpe system, selection of the proper fuel supply mode, requested photographic coverage, and that the safety is stem was operational.
- b. Electromechanical Checkouts. These checks examined, at the test start, the overall control system readiness by individual confirmation of proper openation of each control in the blockhouse.
- preset automatic sequence timer, without actual fuel flow, while recording controlelement actuations on the facility oscillograph. Sequence times of the various elements were measured and adjusted where necessary. The sequence was then repeated, adding a shutdown with the emergency switch to confirm proper emergency switch actuations.

d. Leak Check. These checks provided a game us nitrogen system leak mask at maximum operating pressure for the fuel system, thel vaporizer and condenser log, thel induction system, and the air compressor system.

If TE: The four checklist procedures described above were not performed before each test, but were done when special circumstances, such as a mponent changes, malfunctions, or severe weather, were encountered.

1. Fuel Transfer Procedures

- a. <u>Propellant (fuel) Fill Check Lists</u>. These procedures were provided for transferring liquid fuels from their storage containers into the test stand fuel supply tank. Propane and butane were transferred via their own vapor pressure. The other liquid fuels were transferred from drums by means of an air-motor-iniven pump. It was common to expect up to five separate tests in a day, each if which required approximately $4.6 \times 10^{-3} \, \mathrm{m}^3$ (1 gal) of fuel. Therefore, the fuel supply tank was topped off for each test day. The gaseous fuel system was a aled by simply connecting new pressurized gas cylinders to the supply manifula.
- b. Propellant (fuel) Offload. These transfers from the fuel supply tank were normally returned to the appropriate storage container. Small quantities of propane or butane could also be disposed of through the burn stack. Generally, fuels from the vaporizer/condenser loop remaining in the collector tank were not suitable for recycling and were disposed of as waste. It was necessary to empty the collector tank after every two days of testing.

3. Test Preparations

The Test Preparations Check Lists for instrumentation and test systems were completed concurrently on the day of testing. In the blockhouse, all patchboard connections were completed and instrumentation was setup. An end-to-end instrumentation system calibration was performed. At the test stand, various safety check and facility setups were made: condenser cooling water was turned on, the hydrocarbon analyzer was put in operation, and the hydrogen and air gas pressures were adjusted for the igniter. At the control console, the air compressor was started and the air flow adjusted by means of the air metering valve and the air bypass valve. After the air system temperature and flow were stabilized at the lesired values, the test flame arrester pretest pressure loss was measured and recorded.

The fuel vaporizer heater was activated, and nitrogen purge gas flowed through the heater coils and into the condenser for the preheat cycle. The test stand safety condition was changed from GREEN to AMBER. The fuel supply tank was pressurized with nitrogen up to the desired operating pressure. The vaporizer heater nitrogen purge gas was turned off and fuel flow was metered at a low level. The fuel flow was increased up to the desired test condition as the vaporizer heater reached the operating temperature.

Final visual checks were made of the test stand area, and the ignition completion key switch was turned on. All operating personnel evacuated the test stand area and its safety condition was changed to RED.

4. Blockhouse Preparation

brockhouse preparation because with a weather station outlinear in formation of wind velocity and direction and the focal bar metric pressure. Therefore no le circuits for isnition and emergency shutdown functions were armed and each of the nificant panel switch and its position confirmed. With all test personnel of their operating positions, the test conditions were reviewed and confirmed. In pretest instrumentation calibration was recorded and the countdown procedure was begun.

5. Countdown

A typical "countdown" procedure follows:

- (1) An announcement was made over the public address system to alert percentage in the general area that a detonation may occur. Generally, the detonation noise was very intense and sharp, capable of creating an indirect hazard. A horn signal was also sounded.
- (2) The IDAC tape, EDAC tape, printer, and oscillograph were turned off to a SLOW SPEED.
- (3) The hydrocarbon analyzer purge was turned OFF, allowing the analyzer to sample the fuel/air mixture flowing through the exhaust-burn stack.
- (4) The fuel mixer valve was changed to the RUN position, allowing fuel to flow to the test piping for the first time in the test sequence. The burn-stack-purge valve was opened to sweep out combustible gases from the collector tank vent line. The oscillograph was turned OFF.
- (5) As the fuel/air mixture traveled through the facility piping and into the flame chamber, the hydrocarbon analyzer responded with a steadily increasing signal. The countdown timer was then stopped for a HOLI period, while fuel flow and air flow were confirmed or adjusted, if necessary. During flame chamber testing, the time required for the mixture ratio of chamber exhaust gas to reach the desired level range: from 2 to 27 minutes due to differences in chamber temperature, fuel density, and flow-through characteristics in the test chamber.
- (6) When the COUNTDOW... resumed, the IDAC tape, EDAC tape, and printer were switched to CONTINUOUS MODE and the oscillograph and movie camera were turned ON. The vaporizer heater was turned OFF (flame chamber tests only) to prevent electrical switching noise on the data traces during the test. The high-frequency FM tape recorder was turned ON.
- (7) The hydrocarbon analyzer purge was turned ON, again isolating it from the test system to protect it from possible pressure pulse damage.
- (8) Valves were actuated to the CLOSED position to isolate the low-pressure transducer from possible pressure pulse damage.

- The lemiter was ARMFD by a time lelawitch and the oscillowraph was ewit med to HIGH CHIED.
- The sequence timer was turner N. This caused the imiter to fire for 30% mc. For flame chamber touts, the hydr gen and air valves for the imiter were opened and the transformer energizing the spark plug was a worst simultaneously. Actual duration of the flame was 150 to 200 ms. For mustained burning tests, only the transformer energizing the spark was to be igniter was a world for 300 ms.
- At the end of the decired test time, the test was terminated by operating the EMERGENCY CUTOFF switch. For flame chamber tests, this engaged five seconds after ignition. For sustained burning tests, this sources thirty minutes after ignition or when flame penetration occurred. The EMERGENCY CUTOFF switch triagreed the following events: fuel mixer valve was switched from RUN to CONDENSE position, vaporizer parse was turned ON, vaporizer heater was turned OFF (sustained burning tests only), fuel tank outlet valve (liquid) was CLOSED, and fuel owinger utilet valve (gaseous) was CLOSED.
 - The igniter was UNARMED, the oscillograph changed to LOW SPEED, and the high-frequency tape turned OFF.
- The fuel metering valve was CLOSED and the movie camera was turned CFF.
- Fact capply tank pressure transducers were vented and a posttest calibrate was performed on the instrumentation.
- Fuel supply tank pressure transducers and the test arrester pressure transducers were reopened to the test system and all instrumentation was turned OFF.
- (16) Compressor air flow was maintained to purge residual fuel and combustion by-products from the test piping.

A. Esttest

The p sttest procedure included a visual inspection of the test stand. The room of the last expected examples assemblies, and replaced discs as required. The posttest flame assemblies pressure less was measured and recorded. Chamber diaphragms were required for repeats of flame chamber tests.

In a repeat test was to be made, the hydrocarbon analyzer was checked out. In the single appreciation Freedure would then be restarted from the point of turning a the single appreciate.

Fig. wing the last test of the day, posttest end-to-end calibration of the last gradentation system was made. Fuel in the induction system was pushed back too end of your tenk and the system thoroughly purged with nitrogen gas.

Immediately after each test, the data recorded on the FM tape recorder was played back onto a quick-look oscillograph at an expanded time scale of 8 to 1. This data told the test conductor that he did or did not get ignition, that the flame arrester quenched the flame, or that the flame penetrated through. If the flame arrester was penetrated and the flame speed was high, a playback record was made of the FM tape data at an expanded time scale of 32 to 1 for greater resolution. These records were then analyzed to determine flame speeds and peak pressure rise data.

SECTION VI

FACILITY CHECKOUT TESTS

A. SUBSCALE FLAME CHAMBER TESTS

A series of tests were made to check out facility systems installer specifically for the flashback flame tests. The initial tests were made in a subscale flame chamber while the full-scale chamber was being fabricated. These tests were a beducted to evaluate the new hydrogen/air spark igniter system, the operating precedures required to fill an enlarged chamber with a combustible fuel air mixture that could be verified with measurements on the total hydrocarbon analyser, the effectiveness of the frangible plastic chamber diaphragms, and, finally, the extent and nature of the problems associated with the flame plume emitted from both ends of the test chamber following the diaphragm rupture.

The subscale chamber shown in Figure 6-1 was made from an existion piece of steel pipe 0.91 m (3 ft.) in diameter and 2.13 m (7 ft.) long. It was mented on supports at the exit end of the facility piping. A commercial Pres-Ver screen-flame arrester housing was installed downstream of the witness section for these check-out tests. Frangible diaphragms made from 6-mil-thick black polyethylene plastic sheeting covered both ends of the chamber. A nominal 15.2-cm- (6-in.-diameter hole in the upstream diaphragm provided entrance for the fuel/air mixture, and a nominal 7.6-cm- (3-in.-) diameter hole in the downstream diaphragm provided the exhaust exit. The gas sampling probe for the hydrocarbon analyser was positioned at the center of the downstream hole. The hydrogen/air spark igniter was mounted on a length of pipe in the center of the chamber, such that the point of ignition was at the axial center line. A high-speed motion picture camera viewed the interior through a window port in the bottom of the flame chamber.

Seven test firings were made in the subscale flame chamber using gasoline air mixtures at an injection equivalence ratio ranging from 1.1 to 1.3 (A/F = 13.29 to 11.24). The injection velocity was 1.52 m/s (5 ft/s) through the 15.2-cm-(6-in.-) diameter piping. Three tests were made with a dual 20-mesh screen arrester installed in the Pres-Vac housing, and four were made with the arrester screens removed. Energetic flames were recorded in the chamber when ignition was made after the hydrocarbon analyser measured an equivalence ratio of 0.7 (A/F = 20.dd) or higher in the exhaust flow. The flames entered the piping on every test where the screen arrester was removed. On the first two tests with the screen arrester installed, the flames were quenched. However, on the last test the flame did penetrate the dual 20-mesh screen arrester and enter the facility piping. The motion picture data from this test showed that the hydrogen/air spark igniter was still burning at the time the propagating flame entered the arrester housing. This would have resulted in excessive flame speed that caused the arrester failure. Posttest inspection revealed no damage to the screen arrester.

Both chamber diaphragms ruptured and burned on all tests. The peak chamber pressure rise recorded ranged from 2.07 to 2.76 kN/m 2 (0.3 to 0.4 psid). The visible flame plume emitted from both ends of the chamber extended for a distance of about 1 m (3.3 ft.). All instrumentation and cabling within this area required flame protective covering. The audible noise associated with diaphragm rupture was minimal. Flashback flame in the facility piping did not produce detonations.

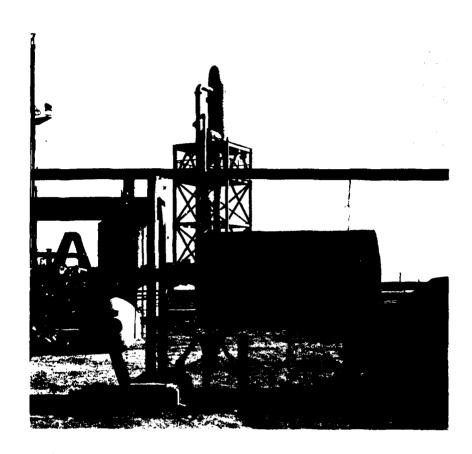


Figure 6-1. Calarage Flame leat Thurver Installation on B-Stand

1. Problem Advantage (1998). Problem (1998).

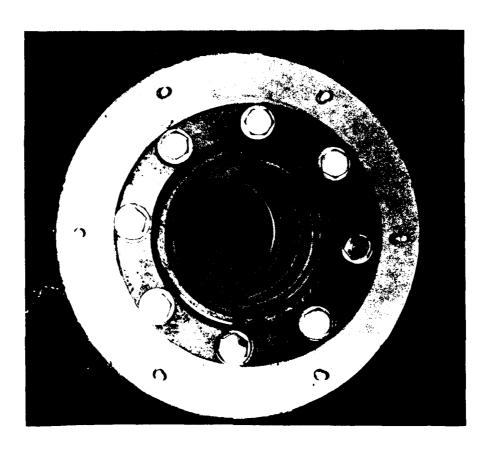
Finder 1-1. These stames increase in administral in the central in like and shown in Finder 1-1. These was among the common for the consequence of the proper for these one was attended to the tests. The infection equivalence easily of the proper air mixture was allowed at the central field with the proper test of the central field of twenty-less than a second fittered the central field were made with the state of the central field of the central field to the central field of the central

The igniter was relocated to the center of the chamber and the test firling were repeated. With this configuration, the flame sensors restricted frame proparation in both the upstream and downstream directions before the chamber liaphrams were blown out. Carculated flame speeds ranged from 1.5 to 4.0 m/s (5 to 15 ft/s) and the flame did not penetrate the screen arrester. All chamber pressure sensors simultaneously recorded a peak pressure rise around 1000 N/m^2 (0.145 psid) just before the diaphragms ruptured.

The igniter was relocated to the upstream position, which places the point source of ignition only 76.2 cm (30 in.) from the downstream face of the screen. One test firing was made with this configuration where the flame penetrated through the screen arrester and into the facility piping. Posttest inspection aid not reveal any damage to the screens. Motion pictures taken of this test showed the ignition sequence and the rapidly expanding spherical flame front. It was estimated that the flame speed was in excess of 15.2 m/s (50 ft/s). This unusually high flame speed was most likely caused by the localized influence of the hydrogen/air spark igniter that was programmed for 2.0 seconds duration. The igniter duration was reprogrammed to only 0.2 seconds (200 ms) on all subsequent tests; this eliminated the high initiation flame speed.

When the igniter was relocated to the downstream position, a 1.52-m- (5-:t.-) diameter aluminum plate was installed to cover the central area of the plastic diaphragm on the flame chamber exit. This flame shield covered about 40% of the total exposed area and delayed the rupture of the diaphragm for a sufficient length of time to allow the flame to traverse the length of the chamber. Motion pictures taken of test firings after this modification showed that the flame propagation path was predominantly in the lower half of the chamber. It is believed that this is caused by gravitational stratification of the fuel/air mixture as it enters the chamber. The 1.52-m/s (5-ft/s) injection velocity is not sufficient to produce turbulent mixing within the large chamber volume. It is, however, representative of the worst-case condition of a fuel storage tank venting vapors on a calm day. The results would be a flammable concentration of fuel vapors collecting in the tank area, causing a very hazardous condition. In the test chamber, the stratified flame produced very inconsistent readings on the flame sensors mounted along the horizontal center line. To correct this situation, the flame sensors were relocated along the top center line, where the field of view looking down into the chamber included the low level flames. The resulting flame speed measurements were much more consistent.

The flame screen assembly, which is mounted in the center of the Pres-Vac housing, was 20.3 cm (8 in.) upstream of the exit flamme. In this position, the screen surface was not visible to the motion picture camera and the flashback flame impingement on the surface of the screen could not be photographed. For the last series of checkout tests, the Pres-Vac housing was replaced with a short 15.2-cm- (6-in.-) diameter flammed pipe spool section to provide the adaptor mounting for the screen flame arresters. The screens were installed between two flammes at the pipe spool exit, where they would be in full view of the motion picture camera. Figure 6-2 is a photograph of a single 30-mesh screen arrester mounted in the pipe spool adapter. Two test firings were made with this test assembly using propane and air mixture at an equivalence ratio of 1.1 and a flow velocity of 1.52 m/s (5 ft/s). Both the upstream and downstream igniter positions were used. Flame speeds from 4.5 to 7.62 m/s (15 to 25 ft/s) were recorded and the flame did not penetrate the single 30-mesh screen arrester on either test.



Then $r=1,\ldots, r=r$, and $r=1,\ldots, r$ where r=r . The High specimens of r

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en en la companya de la co started to evaluate the four selected types of flame arresters with one or more of the eight preselected fuels. To reduce the number of possible tests, a standari test condition was established that would use an injection equivalence ratio (1.0 to 1.2) producing the theoretical maximum flame speed for the particular fuel/air mixture in use and an inlet piping flow velocity of 1.52 m/s (5 ft/s). Ignition would be initiated at an equivalence ratio (0.7 to 0.9) well above the lower flammability limit as measured by the total hydrocarbon analyser sampling the mixture flowing in the exhaust-burn stack.

SECTION VII

DESCRIPTION OF FLAME ARRESTER TEST ASSEMBLIES

A. GENERAL

The U.S. Coast Guard has approved the use of both a single 30-mesh screen and the dual 20-mesh screen configuration for screen flame arresters in U.S. flag vessels. Their purpose is the prevention of flame passage from the open deck into cargo canks through vent outlets, ullage ports, hatches, or butterworth plates. The wire cloth material used for these screens must be resistant to the marine environment, i.e., resistant to chemical corrosion and salt water rusting. In addition, the wire material must be resistant to high-temperature exidation in the event an accidental flame impinges on the screen surface for a prolonged period of time.

These requirements served as guidelines for the selection of flame screen arresters to be experimentally evaluated as part of the U.S. Coast Guard funcied portion of this program. The NASA funded portion was directed at evaluating tw generically different types of flame arresters, namely the spiral-wound, crimped metal ribbon, and the packed bed of Ballast rings. These two types of flame arresters have been shown to be very effective in quenching gasoline/air mixture detonations in a piping system, as reported in Reference 2-10. The propagatine flame speeds for detonations were in excess of 1800 m/s (5906 ft/s). It remaited to be demonstrated that these arresters are also effective against flames with speeds in the range of 1.5 to 9.1 m/s (5 to 30 ft/s), and that they remain effective under sustained burning test conditions for periods up to 30 minutes.

B. SINGLE 30-MESH SCREEN ARRESTER

The single 30-mesh screen arrester was made from standard-grade stainless-steel type 316 wire cloth having the following dimensions:

Mesh size: 30×30 per lineal inch Wire diameter: 0.033 cm (0.013 in.) Hydraulic radius: 0.0516 cm (0.0203 in.) Open area: 37.1%

The type 316 stainless-steel wire is highly resistant to chemical corresion and rusting. It will also resist thermal oxidation at temperatures up to 760° C (1400°F). Nichrome wire has a higher thermal oxidation resistance, up to 972° C (1700°F), but is less readily available in wire cloth weaves.

The single screen, with a Vellumoid gasket on either side, was installed between the exit flange of the 15.2-cm- (6-in.-) diameter pipe spool adapter and a bolted-up, slip-on flange used for clamping, as shown in Figure 6-2. The fuel/air mixture flow velocity in the facility piping varies inversely with the cross-sectional flow area, therefore the standard 1.52-m/s (5-ft/s) flow velocity increases to $4.1 \ m/s$ $(13.5 \ ft/s)$ in passing through the 30-mesh screen attached at the end of the pipe.

T. THAT CHARME MARKET ARRESTER

the the all-meen screen arrester was made from standard grade stainless-steel the two wire countries the following dimensions:

 Mach size:
 20 x 10 per lineal inch

 where diameter:
 0.0-1 cm (0.016 in.)

 hyprodulic radius:
 0.0c6 cm (0.034 in.)

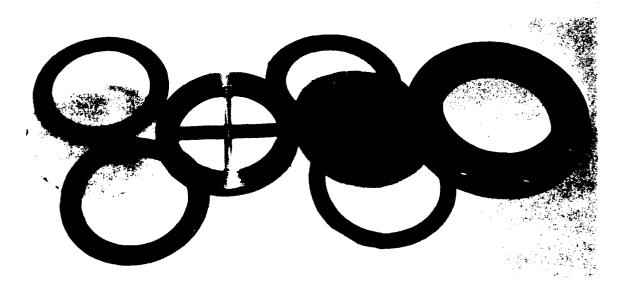
 type area:
 46.2%

the two screens were installed on the pipe spool adapter using the same method as the simple screen, but with the addition of a 0.54-cm- (1.0-in.-) thick spacer separation the two screens. An exploded view of the components in this assembly look which Floure [-1, and the test installation is shown in Figure 7-2. The five all mixture flowing at standard conditions in the facility piping accelerated to 1. The five all fixed passage through the 15.3-cm- (6-in.-) diameter 20-man defrance.

.. Fire A. - W UND, CRIMPED METAL RESEON ARRESTER

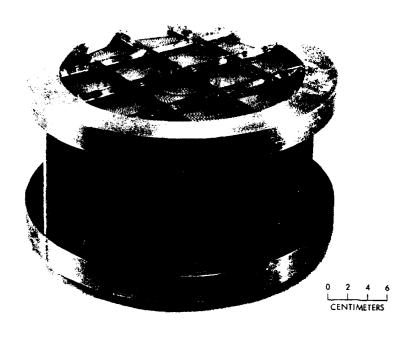
the spiral-wound, crimped metal ribbon arrester was the optimum configuration because the results of the parametric phase of the testing reported in Ferture 2-1). It was made from a commercially available Shand and Jurs spiral-wound, crimped stainless-steel core element having the following dimensions:

| 1 | 1 | 20.5 cm (12 in.) | 20.8 cm (6 in.) | 20.8 cm (6 in.) | 20.8 cm (0.0035 in.) | 20.8 cm (0.063 in.) | 20.8 cm (0.138 in.) |



Pitture Jacobard Text View of Components for a Dual 20-Mesh Screen Arrester

 $\label{eq:controlled} \mathcal{A}^{(1)}(x,y) = \frac{1}{2} \left(\left(\left(\frac{1}{2}





There were the control of the contro

E. PACKED BED OF BALLAST RINGS ARRESTER

The configuration for the packed bed of Ballast rings arrester was all developed during the parametric phase of testing reported in Reference 2-10. It has the following optimized dimensions:

Bed diameter: 25.4 cm (10 in.)

Bed length: 45.7 cm (18 in.)

Bed volume: 3605 cc (1419 cu. in.)

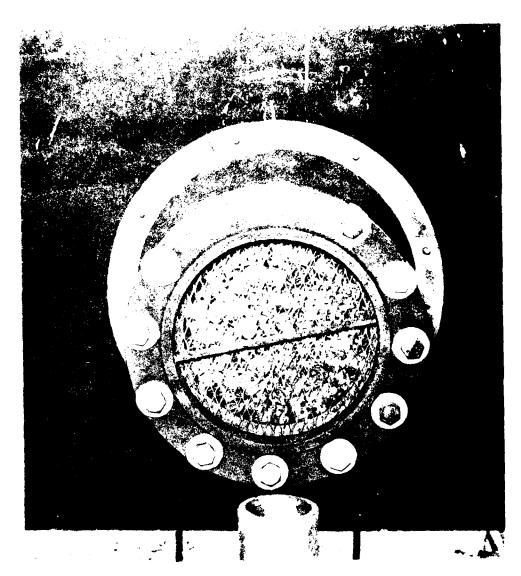
Packing material: aluminum Ballast rings

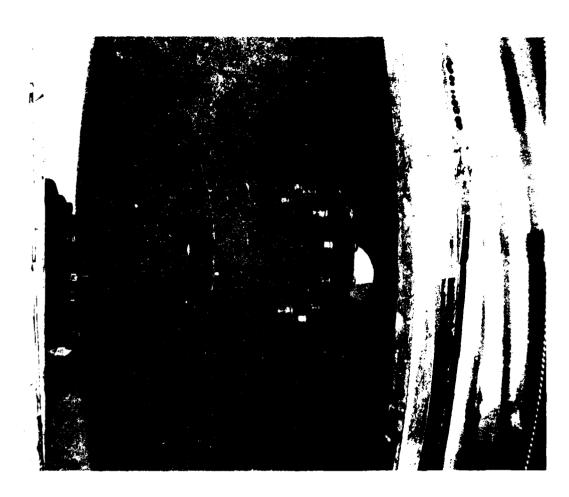
Ring size: 2.54 cm (1.0 in.) in diameter ×

2.54 cm (1.0 in.) long

Open area: 60% (estimated)

The rings were randomly packed in 25.4-cm (10-in.-) diameter flanged pipe housing and held in place with an expanded metal grid, as shown in Figure (-5. A flanged concentric pipe reducer, 25.4-cm to 15.2-cm (10-in. to 6-in.) diameter, adapted the inlet end of the arrester housing for installation on the exit of the facility piping as shown in Figure 7-6. The estimated fuel/air mixture flow velocity through this arrester at the standard test condition was around 0.9 m s (3.0 ft/s).





January II VIII

FIA HEATE FLATE AREESTILE TO USE

A. THOU IDEARAN . TIT

nestert program if roomeer flame arresters if the state of the collections presented in Figure 8-1. After the desertion of the flame arrester to the collection, the first consenior test design was performed in the flame to flame flame the flame of the flame the flame that the flame flame that the flame flame flame the flame
The sec of screening test series was performed to evaluate the screening flame agreeter and journation(s) from the first series, using an ethylene air mixture because it had the highest probable flame speed of the representative bulk care funds. In the upstream and iswastream igniter positions were assumed up no ampletion of the ethylene air mixture tests, near out, of the screen flame agreeters was to be selected for additional testing with the six alternate types of fixed fair mixtures. The selection of agreeter and instantions was made to the U.C. Cast Guard, based upon the test results and the real amendations provides by Fig.

Additional evaluation test series were made in the flame chamber with the selected arrector configurations using the following representative to k carriagles: It aretaidehyle, (2) butane, (5) ethyl ether, (4) gas line, (5) methyl alphin, and (6) to been. The igniter position was relected to produce the most severe flashback flame propagation condition as determined from the measured flame special edvancing toward the face of the test arrector in the first two screening test series.

A final evaluation test series was made using the successful arrester a national from the previous testing to evaluate their heat-up and quanching capabilities in the sustained flame facility. The flame from a proposal mixture at the standard test condition was stablized on the downstream face of the arrester for a period of 30 minutes. These tests were used to determine if the arrester can addition to function after reaching an elevater stangestate, a sheark temperature without structural remark. A single test for the full downtion of 60 minutes, without a flame penetration, was sufficient to demonstrate the successful performance of any arrester configuration. The flame did penetration the test arrester, the test would be repeated to verify the failure.

The two DAJA-frace is flower surrester or filteretians, the opins -words, or insersers offices and the packet being the aminon to task minute, were insert into the or order to wind the section of reening test series. They were evaluated in the

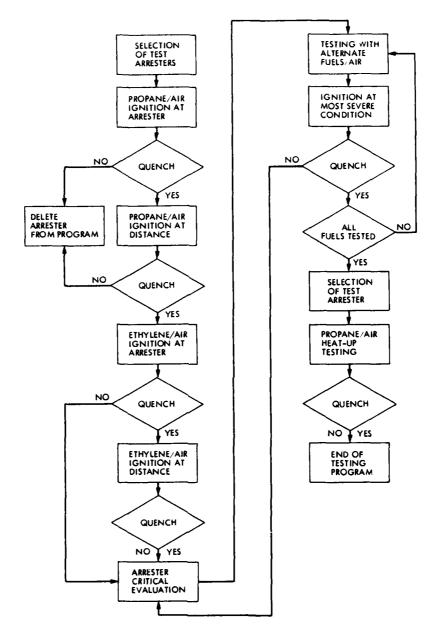


Figure 8-1. Screen-Type Flashback Flame Arrester Test Program Logic Diagram

flacings these chamber with three flacis: if propane, it orbytene, and the flacing. The igniter position was selected for the most determinent of the flat orbits of proposed in mixture and othylene air mixture at the standard test a miltion.

F. BEHAUTIME MIXTURE CORRENING TESTS

The first series of screening tests were made with propose air mixture of the standard test condition where the injection equivalence ratiowas 0.10° (A/F = 13.75). Fill time required to obtain a good combustible mixture in the flame thumber was 600 seconds. The nominal equivalence rational injection was 0.37 (A/F = 13.02) as measured by the total hydrocarbon analyser sampling the field/air mixture in the exhaust-burn stack. Tests were made with the ideal 30-mesh screen arrester and the single 30-mesh screen arrester using both the upstream and 3 whatream igniter positions (Test Configuration Nos. 113 to 116). Successful ignition and combustion was achieved on all test firings. The flash-lack flames did not penetrate either of these screen-type arresters on any test.

The upstream igniter position produced an average flame speed between the plant source of ignition and the arrester (F81-F82, Table 4-1) that measured to an occupance of 13.6 m/s (44.6 ft/s) before it exited the downstream increased to an average of 13.6 m/s (44.6 ft/s) before it exited the downstream end of the flame chamber (F86-F87). Average peak pressure rise in the chamber (F81 to DF87) ranged from 1139 to 974 N/m² (0.165 to 0.141 psid). A plot of the results from these tests is shown in Figure 8-2. Also shown on this plot are the flame speeds in the facility piping that occurred on the last checkout test, when an arrester was not installed. The flame entered the piping (F81-F73) at the modes of 15 ft's) and accelerated up to 18.9 m/s (62.0 ft/s) at the facility inject arrester (F21-F12). A tabular summary of averaged flame speed data and peak pressure rise data is presented in Table 1-1. A tabular summary of all steady-state data is presented in Appendices C and D.

The average flame speeds recorded in this test chamber when using the downstream igniter position (Test Configuration Nos. 114 and 115) were more uniform and lower in value, as shown in the data plot, Figure 8-3. A maximum flame speed fold m/s (18.8 ft/s) occurred just upstream of the igniter (F86-F87). The flashback flame speed propagating against the direction of flow (upstream) was mly 3.0 m/s (9.8 ft/s). This is about one half the speed obtained using the apstream igniter position. Peak pressure rise data were also more uniform and which the speed obtained using the apstream igniter position.

The results of these first screening tests indicate that both the dual 30-mesh screen arrester are effective in quenching flashback flames with a nominal flame speed up to 6.3 m/s (30.7 ft/s). The more severe test condition in the flame chamber is produced when the igniter is located in the upstream position. The flame speed data obtained from the motion picture films corroborate these test results. It was apparent in the films that the degree of intensity (brightness) in the propagating flame from the correlator to the regions of optimum fuel/air mixture ratio and higher levels of the light data turnulence. When the upstream igniter position was used, a bright ball:

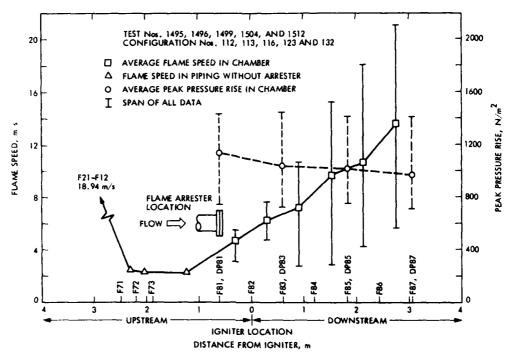


Figure 8-2. Propane/Air Mixture Using Upstream Igniter Position Test Results

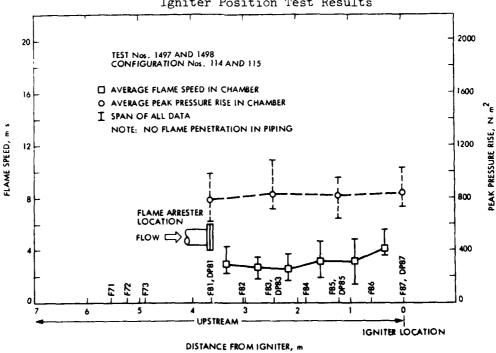


Figure 8-3. Propane/Air Mixture Using Downstream Igniter Position Test Results

of theme eads be seen accelerating upstream through the center core flow of the plane of the fact air mixture as it expanded from the facility pipins. The fact, air mixture as it expanded from the facility pipins. The fact, air mixture rations the expanding plane became stratified by gravitational effects: the heavier hydrocarb in vapors settled to the bottom of the test chamber. When the dismission is indicated mainly in the lower half of the chamber with a very or at and lifthesed flame front moving relatively slowly upstream. This figure from the increased in brightness and accelerated in speed as it progresses up the board to the test arrester installed on the facility pipins.

. FIRYLEDECAIR MIXTURE SCREENING TESTS

The second series of screening tests were made with an ethylene/air mixture of the chandard test conditions. The injection equivalence ratio was 1.15 (A/F = (0.86)) for maximum flame speed. Fill time required to charge the flame chamber with a combustible mixture of this gaseous fuel was reduced to 400 seconds. The number equivalence at the time of ignition was 0.70 (A/F = 21.1). Both the dual (0.0)-mesh screen arrester and the single (0.0)-mesh arrester were used in these tests with upstream and downstream igniter positions (Test Configuration Nos. 117 to (0.0)).

A problem started on the first test when it was discovered that a sustained flame developed inside the exhaust-burn stack piping during the chamber filling operations. It is believed the flame originated from the natural gas fired burner at the top of the stack. Once the ethylene/air exhaust reached a flammable mixture level, a flashback flame from the burner impinged on the exit arrester. The relatively high flame speed of the ethylene/air mixture and the low flow velocity at this location allowed the flame to penetrate into the core of the arrester. It heated the stainless-steel crimped ribbon up to the spontaneous ignition temperature (490°C) for ethylene fuel. At this point, the flame passed through the exit arrester, propagated up the piping, and held on the downstream face of the inlet arrester. Other than blistering the paint on the outside of the piping, this caused no structural damage.

In the inlet arrester of the exhaust-burn stack had a core element made of spiral-wound, crimped aluminum ribbon. It was four times as long as the exit arrester, 15.2 cm (6 in.) compared to 3.8 cm (1.5 in.), and approximately the same diameter. This larger mass of metal, having higher heat capacity, apparently prevented the lean ethylene/air flame from penetrating through the inlet arrester. Consequently, the exit arrester was replaced with a unit similar to the inlet arrester. The results indicated no further incidents of sustained flames in the exhaust-stack piping and the test program to evaluate screen-type arresters using ethylene/air mixture flames continued.

The average flame speeds recorded in the flame chamber when using the downstream igniter position (Test Configurations No. 119 and 121) ranged from 7.8 m/s (25.6 ft/s) at the igniter (F86-F87) to 4.4 m/s (14.4 ft/s) at the arrester (F81-F82). The average peak pressure rise in the chamber was 931 N/m² (0.135 psid). A plot of the test results are shown in Figure 8-4. Both types of screen flame arresters were successful in quenching these ethylene/air mixture flashback flames.

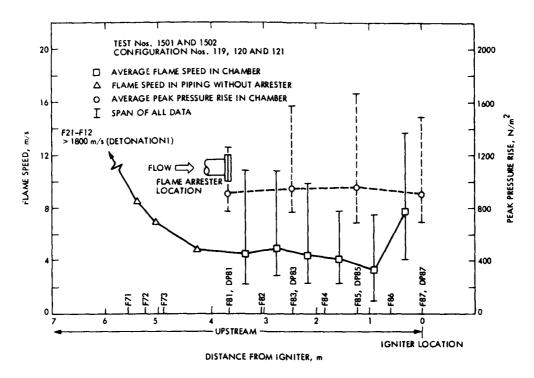


Figure 8-4. Ethylene/Air Mixture Using Downstream Igniter Position Test Results

When the arrester was removed from the end of the facility pipe, the flame entered the pipe (F81-F73) at a speed of 4.9 m/s (16.1 ft/s) and accelerated to a detonation at the inlet arrester (F21-F12) with speeds in excess of 1800 m/s (5905 ft/s). The detonation did not produce any damage to the test facility systems.

The average flame speeds recorded in the flame chamber when using the upstream igniter position (Test Configuration Nos. 117, 118, and 122) ranged from 6.6 m/s (21.6 ft/s) at the arrester (F81-F82) to 16.3 m/s (53.5 ft/s) at the downstream chamber exit (F86-F87). The average peak pressure rise in the chamber was 1102 N/m² (0.160 psid). A plot of the test results are shown in Figure 8-5. The single 30-mesh screen arrester was successful in quenching all flashback flames, whereas the dual 20-mesh screen arrester failed to quench any of the flashback flames in three test firings. The flame that penetrated through the arrester screen housing decelerated briefly to 3.9 m/s (12.8 ft/s) in the facility piping (F81-F73), and then quickly accelerated to a detonation before reaching the facility inlet arrester (F21-F12). Posttest inspection of the screens following each flame penetration did not reveal any damage to the screen wire that could have caused this failure.

The results of the second screening tests indicate that the single 30-mesh screen arrester is effective in quenching flashback flames with nominal flame speeds up to 6.6~m/s (21.6~ft/s). The dual 20-mesh screen arrester is not effective at this higher flame speed, and the limiting flame speed will have to

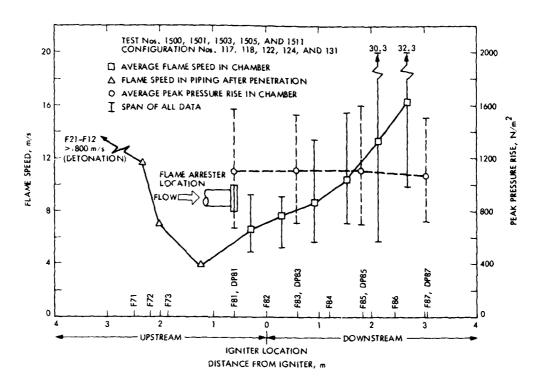


Figure 8-5. Ethylene/Air Mixture Using Upstream Igniter Position Test Results

be determined from additional tests. The upstream igniter position again resulted in the more severe test conditions when using ethylene/air mixture in the flame test chamber. Photographic data of flame speeds taken from the motion picture films corroborated these test results. In accordance with the logic diagram, Figure 8-1, the follow-on alternate fuels tests were limited to using the upstream igniter position only.

D. GASULINE/AIR MIXTURE TESTS

The first series of alternate fuels tests were made with a gasoline/air mixture at the standard test condition. The injection equivalence ratio was 1.10 (A/F = 17.29) for maximum flame speed. Time required to fill the test chamber varied depending on the ambient temperature, but averaged around 900 seconds. The nominal equivalence ratio for ignition was 0.70 (A/F = 20.89). Tests were made using the dual 20-mesh screen arrester, the single 30-mesh screen arrester, the spiral-wound, crimped stainless-steel ribbon arrester, and the packed bed of aluminum Ballast rings arrester (Test Configuration Nos. 125 to 130). All tests were made with the igniter in the upstream position.

The average flame speed between the igniter and the downstream face of the test arresters (F81-F82) was 4.22 m/s (13.3 ft/s). The highest average flame speed was measured just downstream of the igniter (F82-F83) at 6.01 m/s

(19.7 ft/s); from there it decelerated to only 2.92 m/s (9.6 ft/s) at the flame chamber exit (F86-F87). Average peak pressure rise in the chamber was around 1018 N/m^2 (0.148 psid). Without any arrester installed, the flashback flame entered the facility piping at 2.00 m/s (6.6 ft/s) and propagated upstream reaching a speed of 5.44 m/s (17.8 ft/s) at the facility inlet arrester (F21-F12). A plot of the results from these tests is shown in Figure 8-6.

The dual 20-mesh screen arrester, the single 30-mesh screen arrester, and the crimped ribbon arrester were all successful in quenching the flashback flames from the gasoline/air mixture. The packed bed arrester, in the original test configuration (No. 129), was unsuccessful in quenching the first three firings. Flame sensor data actually recorded an acceleration in flame speed during passage through the bed of rings, possibly caused by induced turbulence. A single 30-mesh screen was inserted between the downstream face of the bed and the retainer grid as shown in Figure 8-7. This test configuration (No. 130) was retested using the gasoline/air mixture, propane/air mixture, and ethylene/air mixture. It proved to be successful in quenching the flashback flame from all three fuel/air combinations. During the testing with ethylene/air mixtures, there was evidence of slight pressure spiking in the facility piping 25 seconds after ignition and concurrent with the lean blowout of the flame holding on the downstream face of the arrester. Posttest inspection of the arrester revealed no damage to the screen wire, but there was discolorationindicating that the impinging ethylene/air flame had heated the screen above 550°C (1022°F).

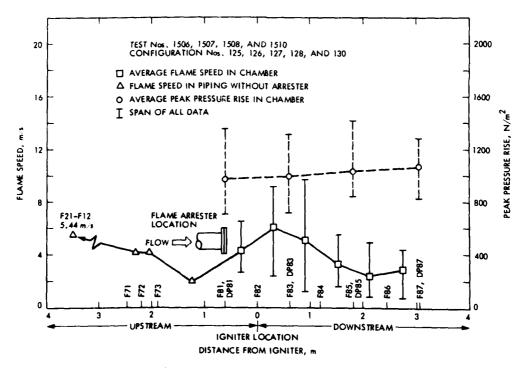
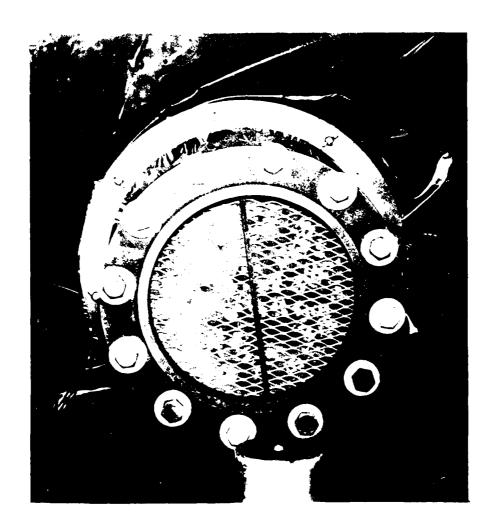


Figure 8-6. Gasoline/Air Mixture Test Results



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E. METHANOL/AIR MIXTURE TESTS

The second series of alternate fuels tests was made with methanol/air mixture at standard test conditions. The injection equivalence ratio was 1.01 (A/F = 6.41) for maximum flame speed. Time required to fill the test chamber averaged 1060 seconds, because of the cold ambient temperatures and the low valatisity of methanol. The nominal measured equivalence ratio at ignition was 0.69 (A/F = 9.38). Tests were made with the dual 20-mesh screen arrester and the single 30-mesh screen arrester using the upstream igniter position (Test Configuration Nos. 133 to 135).

The average flame speed between the igniter and the downstream face of the test arresters (F81-F82) was 4.35~m/s (14.3 ft/s). The highest average flame speed measured just downstream of the igniter (F82-F83) was 5.52~m/s (18.1 ft/s). Two flame sensors at the exit of the flame chamber (F86 and F87) were inoperative due to weather conditions. The average peak pressure rise in the chamber was $831~\text{sl/m}^2$ (0.120 psid). Without an arrester installed, the flashback flame entered the facility piping with a flame speed of only 2.19 m/s (7.2 ft/s), and was unable to propagate upstream through the facility piping. A plot of the results from these tests is shown in Figure 8-8. Both the dual 20-mesh screens arrester and the single 30-mesh screen arrester were successful in quenching all flashback flames from the methanol/air mixture.

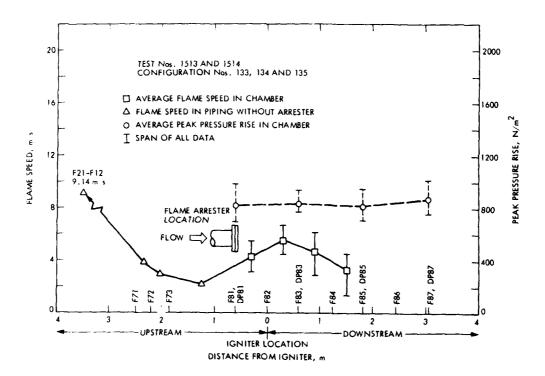


Figure 8-8. Methanol/Air Mixture Test Results

F. TOLUENE/AIR MIXTURE TESTS

The third series of alternate fuels tests were made with toulenclair mix are at standard test conditions. The injection equivalence ratio was 1.05 (A/F = 10.56) for maximum flame speci. Time required to fill the test chamber averaged 1070 seconds. The nominal measured equivalence ratio at ignition was 0.68 (A/F = 19.9). Tests were made with the dual 20-mesh screen arrester and the single 50-mesh screen arrester using the upstream igniter position (Test Configuration Nos. 136 to 138).

The average flame speed between the igniter and the downstream face of the test arresters (F81-F82) was 5.42 m/s (17.8 ft/s). The highest average flame speed measured just downstream of the igniter (F82-F83) was 6.27 m/s (20.6 ft/s); from there it decelerated to only 2.65 m/s (8.7 ft/s) at the flame chamber exit (F86-F87). The average peak pressure rise in the shamber was 668 H m² (0.096 psid), the lowest value recorded for all fuel/air mixtures. Without a flame arrester installed, the flashback flame entered the facility piping with a flame speed of only 0.61 m/s (2.0 ft/s) and was unable to propagate upstream through the facility piping. A plot of the results from these tests is shown in Figure d-9. Both the dual 20-mesh screen arrester and the single 30-mesh screen arrester were successful in quenching all flashback flames from the toluere/air mixtures.

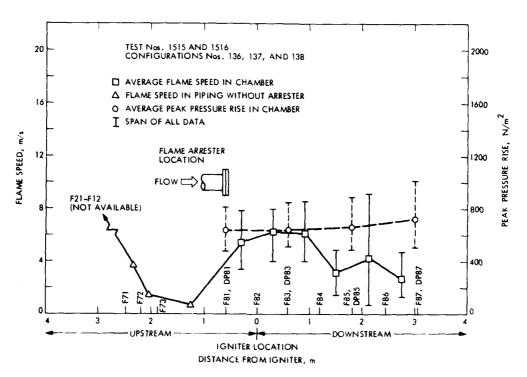


Figure 8-9. Toluene/Air Mixture Test Results

G. PRETEYL ETHER WIR MIXTURE TESTS

The fourth peries of alternate fuels tests were made with diethyl ether/air mixture at spendard test conditions. The injection equivalence ratio was 1.15 (A F = 9.73) for maximum flame speed. Time required to fill the test chember averages about seconds. The nominal measured equivalence ratio at the time of ignition was 0.71 (A/F = 15.8). Tests were made with the dual 20-mesh screen arrester and the single 30-mesh screen arrester using the upstream igniter position (Test Configuration Nos. 139 to 141).

The average flame speed between the igniter and the downstream face of the test arrester (F81-F82) was 5.61 m/s (21.4 ft/s). The highest average flame speed measured in the center of the chamber (F84-F85) was 11.95 m/s (39.2 ft/s). These flame speeds were the second highest obtained, next to the ethylene/air mixture. The average peak pressure rise in the chamber was 937 N/m² (0.136 psid). Without an arrester installed, the flashback flame entered the facility piping with a flame speed of 2.98 m/s (9.78 ft/s) and propagated upstream accelerating to 59.43 m/s (195 ft/s) at the facility inlet arrester (F21-F12). A plot of the results from these tests is shown in Figure 8-10. Both the dual 20-mesh screen arrester and the single 30-mesh screen arrester were successful in quenching all flashback flames from the diethyl ether/air mixture.

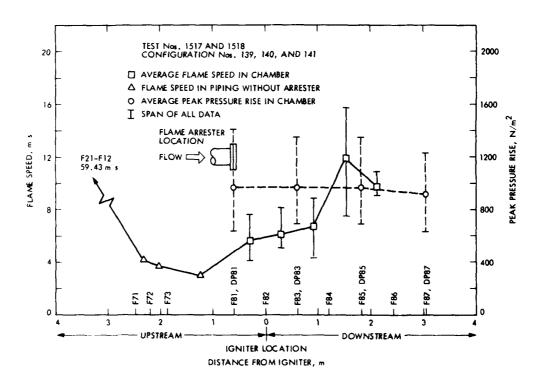


Figure 8-10. Diethyl Ether/Air Mixture Test Results

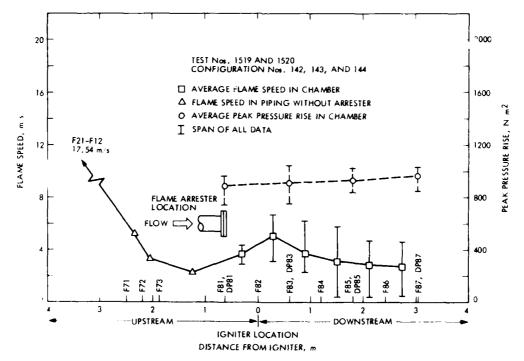
H. BUTANE/AIR MIXTURE TESTS

The fifth series of alternate fuels tests were more with nutaneous mixture at standard test conditions. The injection equivalence ratio was the (A/F = 13.68) for maximum flame speed. Time requires to fill the test smeater averaged 507 seconds. The nominal measured equivalence ratio at the time of ignition was 0.78 (A/F = 19.8). Tests were made with the dual ro-mesh screen arrester and the single 30-mesh screen arrester using the upstream lamited position (Test Configuration Nos. 142 to 144).

The average flame speed between the igniter and the downstream face of the test arrester (F81-F82) was 3.62 m/s (11.9 ft/s). The highest average flame speed measured just downstream of the igniter (F82-F83) was 5.07 m/s (16.6 fter); from there it decelerated to only 2.71 m/s (8.9 ft/s) at the flame chamber exit (F86-F87). The average peak pressure rise in the chamber was 926 $11/m^2$ (0.140 psid). Without an arrester installed, the flacebook flame entered the facility piping with a flame speed of 2.26 m/s (7.4 ft/s) and propagated upstream accelerating to 17.54 m/s (57.5 ft/s) at the facility inlet arrester (F21-F12). A plot of the results from these tests is shown in Figure 8-11. Both the ideal 20-mesh screen arrester and the single 30-mesh screen arrester were suressift. In quenching all flashback flames from the butane/air mixtures.

I. ACETALDEHYDE/AIR MIXTURE TEST

The sixth and final series of alternate fless tests were made with acetaldehydr/air mixture at standard test conditions. The injection equivalence



Fire n-11. Butane/Air Mixture Test Results

ratio was 1.15 (A/F = 6.82) for maximum flame speed. Time required to fill the test chamber averaged 920 seconds. The nominal measured equivalence ratio at the time of ignition was 0.63 (A/F = 12.5). Tests were made with the dual 20-mesh screen arrester and the single 30-mesh screen arrester using the upstream igniter position (Test Configuration Nos. 145 to 147).

The average flame speed between the igniter and the downstream face of the test arrester (F81-F82) was 5.30 m/s (17.4 ft/s). The highest average flame speed measured at the chamber exit (F86-F87) was 12.11 m/s (39.7 ft/s). These flame speeds are about equal to those obtained for the diethyl ether/air mixture. The average peak pressure rise in the chamber was 1102 N/m^2 (0.160 psid), which is the same level obtained with ethylene/air mixture. Without an arrester installed, the flashback flame entered the facility piping with a flame speed of 3.22 m/s (10.6 ft/s) and propagated upstream accelerating to 411 m/s (1348 ft/s) at the facility inlet arrester (F21-F12). A plot of the results from these tests is shown in Figure 8-12. Both the dual 20-mesh screen arrester and the single 30-mesh screen arrester were successful in quenching all flashback flames from the acetaldehyde/air mixture.

J. ARRESTER SELECTION FOR SUSTAINED BURNING TESTS

The tests described above completed the alternate fuel/air mixtures step in the test program logic diagram presented in Figure 8-1. Since both the dual 26-mesh screen arrester and the single 36-mesh screen arrester were successful

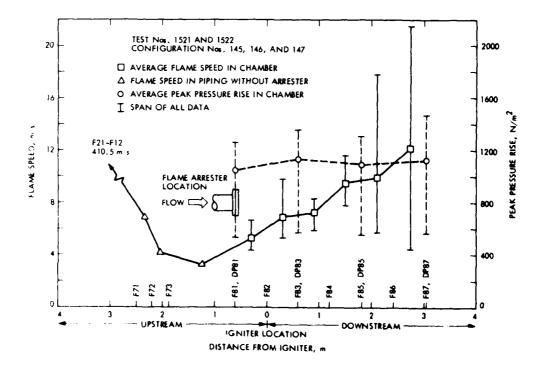


Figure 8-12. Acetaldehyde/Air Mixture Test Results

in quenching all flashback flumes from all the alternate fuel tests, they were both designated by the U.S. Coast Guard for sustained burning tests, along with the crimped ribbon arrester and the packed bed arrester for the NASA project.

SECTION IX

SUSTAINED BURNING ARRESTER TESTS

A. PROPANEZAIR MIXTURE TESTS

The first series of sustained burning tests were made with propane/air mixtures at the standard test condition where the injection equivalence ratio was 1.14 (A/F = 13.75). The duration of testing was planned for 30 minutes to all w sufficient time for the test assembly to reach thermal equilibrium. In the event the flame penetrated through the arrester, the test was terminated as plackly as possible to minimize damage to the facility piping and instrumentation.

The dual 20-mesh screen arrester and the single 30-mesh screen arrester wave-tested in two different test assembly sizes, the original 15.2-cm (6-in.) Hameter and a new 25.4-cm (10-in.) diameter. This was done to evaluate the effects of the fuel air mixture approach velocity and flow-through velocity on the thermal environment at the screens. The spiral-wound, crimped stainless-stee, rith a arrester and the packed bed of Ballast rings arrester were the same on flowration that proved successful in the flashback flame testing. All arresters were instrumented with additional thermocouples (Figure 9-1) to measure thermal builting and to aid in predicting an impending flame penetration when the arrester tengeneture approached the spontaneous ignition temperature of the fuel/air mixture.

The following results are for the propane/air mixture sustaine: crainstests. A tabular summary of the test data is presented in Appendix E.

2. Single 30-Mesh Joreen Arrester, 15.2-cm Diameter

A schematic drawing if this arrester test assembly "Test "indicated in No. 19 presented in Figure 9-1, shows the 1 dation of the theomorphic 10A user to measure the screen temperature. The small sheath-type therm of the was months with spring adding against the upstream face of the screen. This means was user to maintain point contact and to minimize a case it workstarbance. The approximation of the 15. -em- 6-in. - diameter pipe adapter to which was also most 15. Office and the flow-through vesselty in the screen was also most 15. Office and the flow-through vesselty in the screen was also most 15. Office and the flow-through vesselty in the screen was also most 15. Office and the flow-through temperature reached an initial plateau of the 1500°F) after 100 seconds. The temperature continued to increase stowns after the properature of this adding to the properature air nixture aid in the penetrate through the single vessels acreen accorder. A plot of the results is presented in Figure 9-1. Institute discontration of the screen revealed in armage or flame erisi in and only slight discontration of the wire mesh.

d. Duan Co-Mesn Arrester, 15.1-cm Piameter

A schematic frawing of this test assembly (Test Configuration No. .59%, presented in Figure 9-2, shows the location of the thermocouples (T8A and TSF wheel to measure the two screen temperatures. The approaching fix w velocity in the .5. - cm- (6-in.-) liameter pipe was 1.5 m/s (5.0 ft/s) and the flow-through velocity in the screens was 3.3 m/s (10.8 ft/s). Temperature on the downstream screen (75A) reached an initial plateau of 92° C (198°F) after 120 seconds of operation and



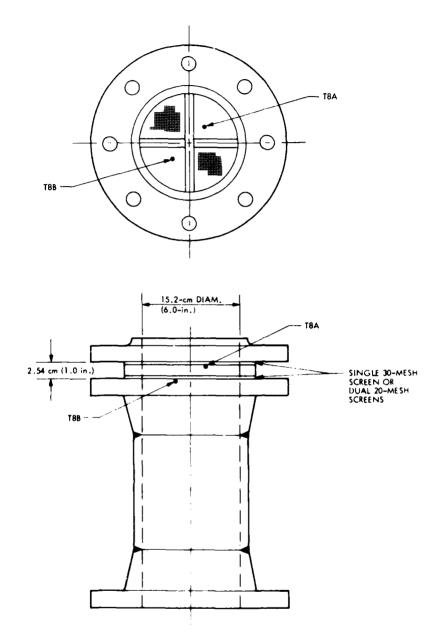
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Figure 9- . Coreen-Type Arrester Test Assembly, 15.2-om. limeter, Cobematic Irawing

that the screen temperature varies inversely with the flow-through velocity, as would be expected. The ne-thir: wer flow-through velocity of this larger screen surface resulted in a ak-back temperatures three times higher than the smaller screen notes above in largeraph A-1 of this section. The propane/air

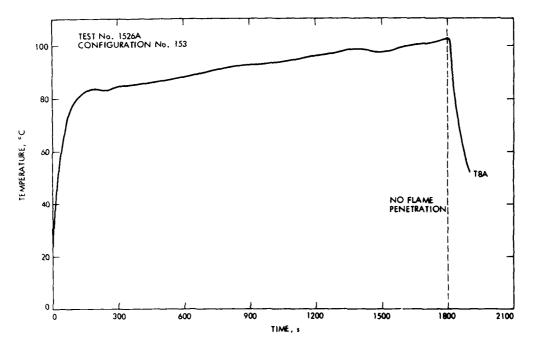


Figure 9-3. Single 30-Mesh Screen Arrester, 15.2-cm Diameter, Propane/Air Mixture Sustained Burning Test Results

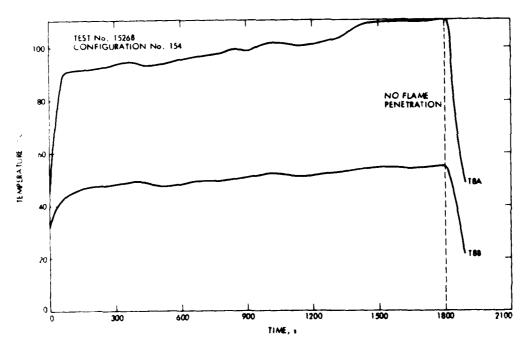


Figure 9-4. Dual 20-Mesh Screen Arrester, 15.2-cm Diameter, Propane/Air Mixture Sustained Burning Test Results

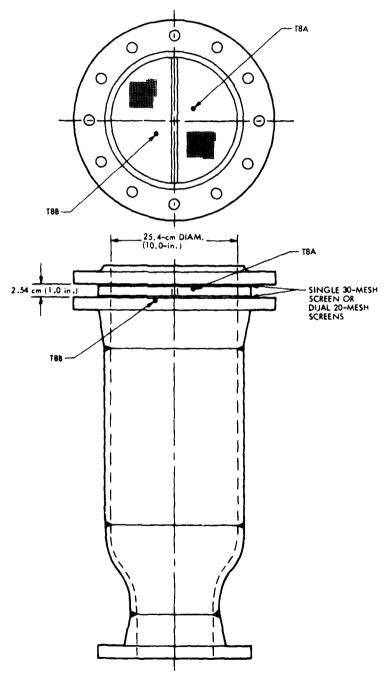


Figure 9-5. Screen-Type Arrester Test Assembly, 25.4-cm Diameter, Schematic Drawing

mixture flame did not penetrate through this single 30-mesh screen arrester. A plot of the test results is presented in Figure 9-6. Posttest inspection revealed only slight discoloration of the wire mesh over about 60% of the surface area as shown in Figure 9-7.

4. Dual 20-Mesh Screen Arrester, 25.4-cm Diameter

A schematic drawing of this arrester test assembly (Test Configuration No. 156), presented in Figure 9-5, shows the location of the thermocouples (T8A and T8B) used to measure the two screens' temperatures. The approaching flow velocity in the 25.4-cm- (10-in.-) diameter pipe was 0.56 m/s (1.8 ft/s) and the flow-through velocity in the screens was 1.21 m/s (3.96 ft/s). The temperature on the downstream screen (T8A) reached an initial plateau value of 160°C (320°F) after 120 seconds of operation and then increased to a nominal value of 190°C (374°F) for the remaining 30 minutes of operation. The upstream screen temperature (T8B) reached 60°C (140°F) after 60 seconds and then slowly increased to 70°C (158°F) by the end of test. The propane/air mixture flame did not penetrate through this dual 20-mesh screen arrester. A plot of the test results is presented in Figure 9-8.

The maximum temperature for this 20-mesh screen arrester assembly was expected to be higher than that measured on the similar sized 30-mesh screen arrester, because of the lower flow-through velocity. Posttest inspection revealed that the thermocouple (T8A) was making poor contact with the screen surface and was sociated in an area of low temperature, as indicated by the flame impingement pattern on the screen. There was no damage to the screens other than a discoloration covering about 60% of the flow area on the downstream wire mesh. A posttest photograph of the 20-mesh screens and spacer is presented in Figure 9-9.

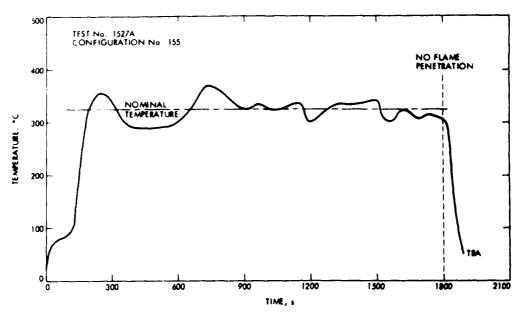


Figure 9-6. Single 30-Mesh Screen Arrester, 25.4-cm Diameter, Propane/Air Mixture Sustained Burning Test Results

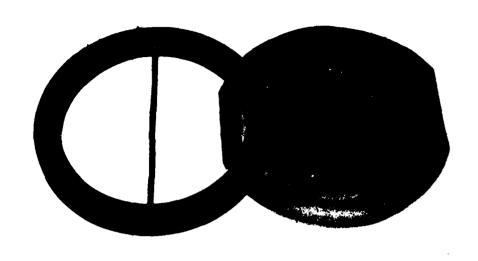


Figure 9-7. Single 30-Mesh Screen Arrester, 25.4-cm Diameter, Posttest

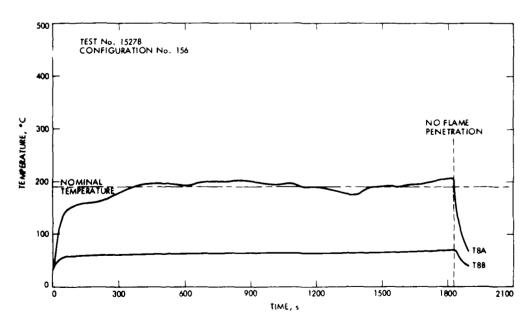


Figure 9-8. Dual 20-Mesh Screen Arrester, 25.4-cm Diameter, Propane/Air Mixture Pustained Burning Test Results

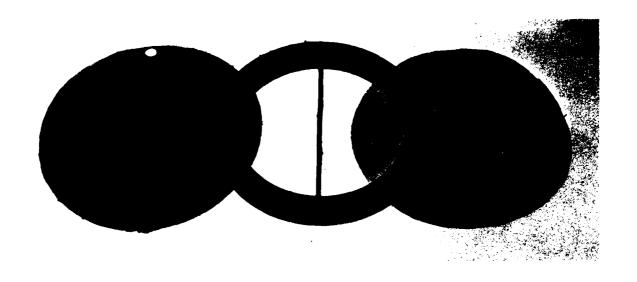


Figure 9-9. Dual 20-Mesh Screen Arrester, 25.4-cm Diameter, Posttest

5. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester

A semember drawing of this arrester test assembly (Test Configuration U. 149), presented in Figure 9-10 shows the location of the six thermocouples TEA to THE queed to measure the crimped ribbon core element temperature. The egreeneding flow velocity in the 30.5-cm- (12-in.-) diameter pipe was 0.39 m/s (1... this) and the flow-through velocity in the crimped ribbon core element was 0.45 m/s (1.46 ft/s). Temperature at the downstream center of the core Floment (Thi) reached a maximum value of 1000°C (1832°F) after 900 seconds disperation and then slowly decreased to 930°C (1706°F) at the end of the to minutes (1800 seconds). The sustained flame had to be burning inside the core element to produce this high temperature, which is considerably above the spontangent limition temperature of 504°C (940°F) for the propane/air mixture. The renter of the core element (T8B) reached this spontaneous ignition temperature Just 10 seconds before test termination. It appears that the sustained flame was initially contined to the center portion of the downstream face, and after It. I recently of operation, the flame had expanded to the outer perimeter (T8A). The receases air mixture flame did not penetrate through this spiral-wound, wrim: ed stainless-steel arrester during the 30-minute test duration. However, the core element had not reached a state of thermal equilibrium and there is a nationable evidence of continuing flame propagation into the core. It is milto likely that the arrester would have eventually failed. A plot of the test regular is a resented in Figure 9-11.

restrict imagestion of this arrester test assembly revealed some minor remove to the core element in the form of distortion and discoloration to the condition-steed ribben windings. The retainer grid was also distorted from rectricts thermal expansion and some grid elements were broken at the weld defines. As a test photograph of the downstream end of the arrester assembly in presents in Figure 9-18.

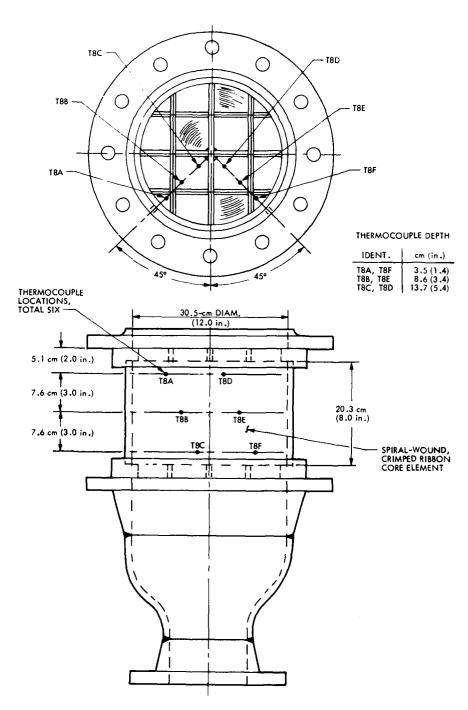


Figure 9-10. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Test Assembly Schematic Drawing

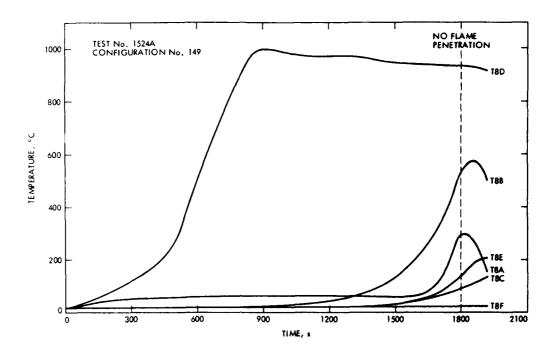


Figure 9-11. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Propane/Air Mixture Sustained Burning Test Results

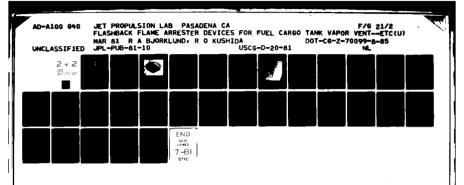
6. Packed Bed of Aluminum Ballast Rings Arrester

A schematic drawing of this arrester test assembly (Test Configuration No. 151), presented in Figure 9-13, shows the location of seven thermocouples (T8A to T8G) used to measure the temperature in the bed of rings and on the single 30-mesh screen retainer. The approaching flow velocity in the 25.4-cm-(10-in.-) diameter pipe was 0.56 m/s (1.8 ft/s), the flow-through velocity in the bed of rings is estimated at 0.94 m/s (3.1 ft/s), and the flow-through velocity in the 30-mesh screen was 1.5 m/s (4.9 ft/s). Temperature of the screen (T8G) reached the nominal value of 350°C (662°F) after 200 seconds of operation and held fairly steady for the 30 minutes duration. The temperatures at the top of the bed (T8A and T8D) increased slightly to a maximum of 125°C (257°F) due to radiation only; very little conductive and no convective heating was possible. The lower part of the bed remained at the nominal mixture inlet temperature of 50°C (122°F). The propane/air mixture flame did not penetrate through the 30-mesh retainer screen on the packed bed of rings during the 30-minute test duration. A plot of the test results is shown in Figure 9-14. Posttest inspection revealed only a slight downstream bowing and discoloration of the retainer grid and screen as shown in Figure 9-15.



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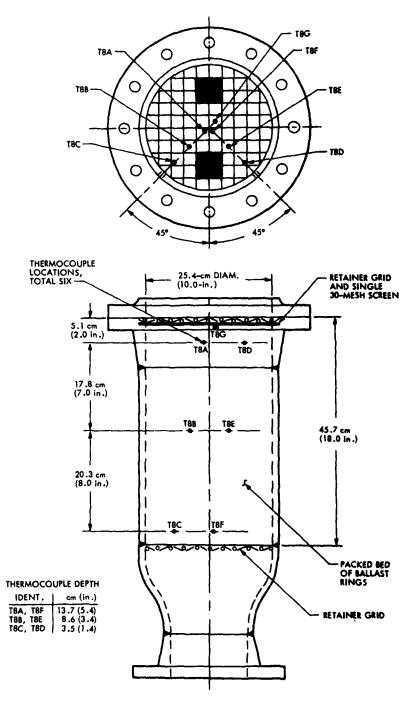


Figure 9-13. Packed Bed of Aluminum Ballast Rings with Single 30-Mesh Screen Arrester Test Assembly Schematic Drawing

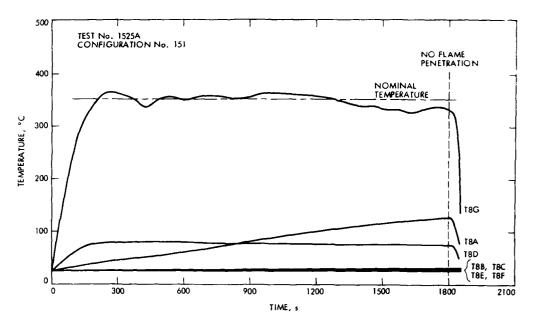


Figure 9-14. Packed Bed of Aluminum Ballast Rings with Single 30-Mesh Screen Arrester Propane/Air Mixture Sustained Burning Test Results

P. ETHYLENE/AIR MIXTURE TESTS

This last series of sustained burning tests were made with ethylene/air mixture at standard test conditions where the injection equivalence ratio was 1.15 (A/F = 12.86). The planned test duration was 30 minutes. Only the two arrester configurations of the NASA funded program were tested: (1) the spiral-wound, crimped stainless-steel ribbon arrester, and (2) the packed bed of aluminum Ballast rings. The USCG funded program did not require sustained burning tests with ethylene/air mixtures because the test conditions were considered to be too severe for screen-type flame arresters.

The following results are for the ethylene/air mixture sustained burning tests. A tabular summary of the test data is presented in Appendix E.

1. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester

This is the same arrester test assembly (Test Configuration No. 150) shown in Figure 9-10. The test flow conditions were the same as those described in Paragraph A-5 of this section. On the first test (No. 1524B) the flame penetrated into the core (T8A and T8E) after only 60 seconds of operation and reached a high temperature of around 900°C (1652°F) at 150 seconds. The flame spread to the outer perimeter of the core (T8B and T8D) increasing this area temperature to



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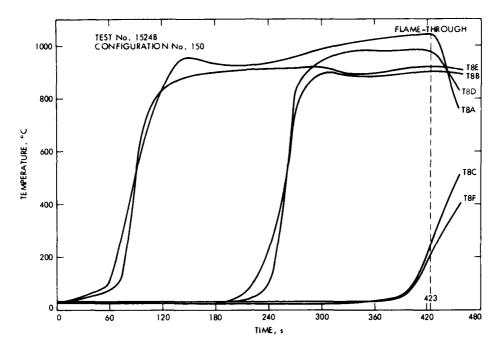


Figure 9-16. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Ethylene/Air Mixture Sustained Burning First Test Results

2. Packed Bed of Aluminum Ballast Rings Arrester

This is the same arrester test assembly (Test Configuration No. 152) shown in Figure 9-13. The test flow conditions were the same as those described in Paragraph A-6 of this section. In the first test (No. 1525B) the temperature on the upstream face of the retainer screen (T8G) increased rapidly, reaching the spontaneous ignition level of 490°C (914°F) after only 35 seconds of operation. Flame penetration occurred at 43 seconds when the screen temperature reached 560°C (1040°F). The bed of aluminum Ballast rings remained at the inlet ethylene/air mixture temperature with only the downstream center of the bed (T8A) receiving any measurable radiation from the sustained burning. Flame penetration through the retainer screen was followed by a detonation in the inlet piping. Flame speeds measured in the witness section, which was just upstream of the test arrester section, were at the detonation velocity of around 1830 m/s (6000 ft/s). This would indicate that the penetrating flame had made the transition from deflagration to detonation within the length of the packed bed arrester. A plot of the test results is presented in Figure 9-18. Posttest inspection of the arrester revealed some distortion and discoloration of the retainer grid and screen assembly caused by internal pressure developed during the detonation.

The above test was repeated at the same test conditions and with the same arrester test assembly. This second test (No. 1525C) resulted in a detonation immediately after ignition. Posttest disassembly and inspection of the packed bed arrester revealed that the screen retainer had been impacted in several

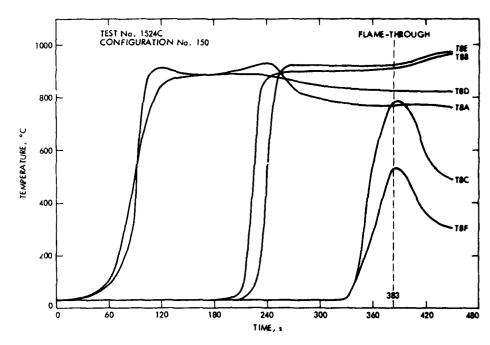


Figure 9-17. Spiral-Wound, Crimped Stainless-Steel Ribbon Arrester Ethylene/Air Mixture Sustained Burning Second Test Results

places by Ballast rings causing punctures as shown in Figure 9-19. The undetected damage to the screen was probably initiated to a lesser extent during the first sustained burning test that resulted in a detonation. These small punctures allowed flame penetration without heat-up on the second test and the subsequent detonation enlarged the holes to the size shown.

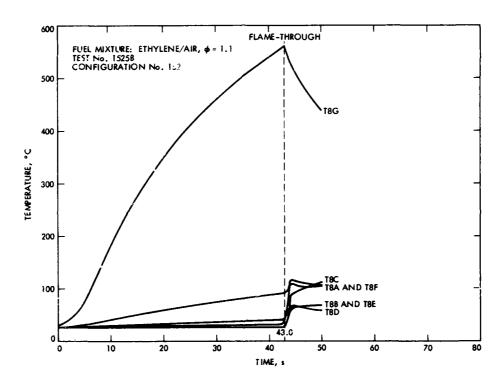
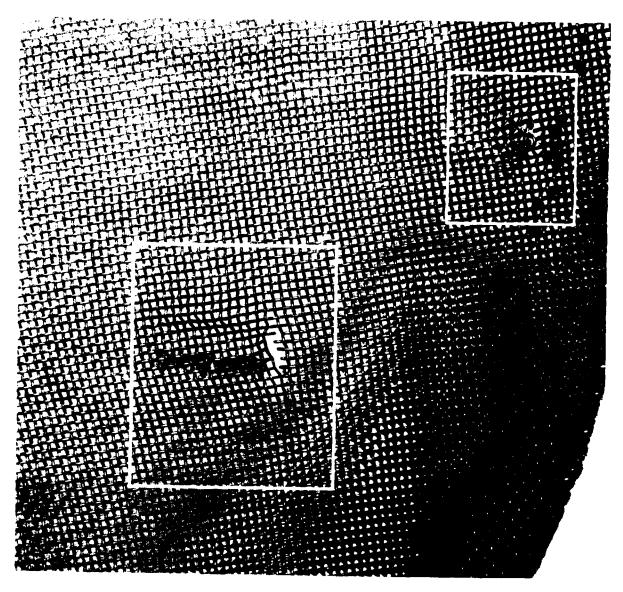


Figure 9-18. Packed Bed of Ballast Rings with Single 30-Mesh Screen Arrester Ethylene/Air Sustained Burning Test Results



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SECTION X

CONCLUSIONS

The following conclusions have been reached from the test results of this experimental evaluation of flame arrester devices in a simulated fuel storage tank vent stack installation discharging eight types of combustible fuel/air mixtures, including: (1) propane, (2) ethylene, (3) gasoline, (4) methanol, (5) toluene, (6) diethyl ether, (7) butane, and (8) acetaldehyde. The test flame arresters were mounted on the end of a 15.2-cm- (6-in.-) diameter pipe vent located in an unconfined one-atmosphere environment. The standard test condition used an injection equivalence ratio from 1.0 to 1.2 to produce the theoretical maximum flame speed for the particular fuel/air mixture in use; the fuel/air mixture temperature ranged from 10 to 38°C (50 to 100°F), and the injet piping nominal flow velocity was 1.52 m/s (5 ft/s).

- (1) An ignition source upstream near the flame arrester and in the center of the exhaust plume produced the highest flashback flame speed for a flame propagating upstream in the direction of the arrester.
- (2) Ethylene/air mixture produced the highest average flashback flame speed of 6.60 m/s (21.65 ft/s), ranging from 4.86 to 10.66 m/s (15.94 to 34.98 ft/s).
- (3) Butane/air mixture produced the lowest average flashback flame speed of 3.62 m/s (11.88 ft/s), ranging from 2.92 to 4.25 m/s (9.58 to 13.94 ft/s).
- (4) Flashback flames from the typical bulk cargo fuels tested will propagate in an open environment, such as the deck of a transport vessel, but will not produce a detonation unless they penetrate an opening leading into a fuel cargo tank.
- (5) The single 30-mesh stainless-steel screen arrester was effective in quenching flashback flames from all eight fuel/air mixtures tested.
- (6) The dual 20-mesh stainless-steel screen arrester was effective in quenching flashback flames from all eight fuel/air mixtures tested except the ethylene/air mixture, where the flame speed was 4.86 m/s (15.94 ft/s) or faster.
- (7) Damage to a screen flame arrester from a puncture, tear, or corrosion that results in holes larger than the original mesh size renders the screen useless in quenching a flashback flame. The damaged screen should be replaced to restore the arrester's effectiveness.
- (8) The spiral-wound, crimped stainless-steel ribbon arrester was effective in quenching flashback flames from the propane, ethylene, and gasoline fuel/air mixtures tested, and would probably quench the other five fuel/air mixtures listed.

- The packed bed of aluminum Ballast rings arrester with single 30-mesh stainless-steel screen retainers was effective in quenching flashback flames from the propane, ethylene, and gasoline fuel/air mixtures tested, and would probably quench the other five fuel/air mixtures listed.
- The packed bed of aluminum Ballast rings arrester without the single 30-mesh screen retainer was not effective in quenching flashback flames from gasoline/air mixtures, and would probably not quench the other seven fuel/air mixtures listed.
- The test configurations for the single 30-mesh screen arrester, the dual 20-mesh screen arrester, the spiral-wound, crimped ribbon arrester, and the packed bed of Ballast rings arrester withstood all flashback flame testing without any structural damage and only slight discoloration from the short duration of flame impingement (approximately seconds).
- The single 30-mesh screen arrester and the dual 20-mesh screen arrester withstood flames from propane/air mixtures for 30 minutes without structural damage and only slight discoloration of the screen wire. The fuel/air mixture flow velocity through the openings in the screen ranged from 1.2 to 4.1 m/s (3.9 to 13.5 ft/s), depending on the size of the arrester test assembly. In each configuration, the screens reached a condition approaching thermal equilibrium after approximately <00 seconds where the temperature was well below the spontaneous ignition temperature for the propane/air mixture. It is concluded that the sustained burning conditions on these arresters could have continged for an indefinite period of time.
- The equilibrium temperature on the surface of a screen flame arrester at sustained burning conditions is a function of flow velocity of the fuel/air mixture passing through the screen; the lower the velocity, the higher the equilibrium temperature. It is possible that at very low flow-through velocities the temperature of the screen would increase to the spontaneous ignition temperature of the fuel and the flame could penetrate the screen arrester.
- The spiral-wound, crimped ribbon arrester withstood flames from the propane/air mixture for 30 minutes. During this time, the flame propagated into part of the depth of the core element, causing distortion and discoloration of the stainless-steel ribbon. Thermal equilibrium within the core element was not achieved during the 30 minutes of testing as the temperatures measured inside the ribbon windings continued to increase above the spontaneous ignition temperature for propane/air mixtures. It is concluded that the flame would have eventually penetrated the arrester, given sufficient time. Sustained burning from the ethylene/air mixture did penetrate through this arrester on two tests of 423 and 383 seconds. Therefore, the ability of this type of flame arrester to withstand sustained burning is highly dependent on the flame speed and the spontaneous ignition temperature of the fuel/air mixture.

(15) The packed bed of Ballast rings arrester with a single 30-mesh screen retainer withstood flames from the propane/air mixture for 30 minutes. The results were very similar to those obtained from the single 30-mesh screen arrester, and it is apparent that the bed of rings has little or no influence on the performance of this arrester configuration. Sustained burning from the ethylene/air mixture dispenetrate through this arrester in only 43 seconds on one test, restricting in a deflagration-to-detonation transition within the bed of rings. The retainer screen was damaged by impacts from the bed of rings, and this damage allowed the flame to penetrate immediately after ignition on a repeat test. It is concluded that the packed bed of rings arrester with a single 30-mesh screen is no more effective than a single 30-mesh screen in withstanding and quenching flashback flames.

SECTION XI

RECOMMENDATIONS

Based upon the results of this test program, the following resommendation are made regarding the selection and installation of flame arresting devices in fuel storage tank vent stacks in a marine environment:

- (1) Based upon flame quenching capability, structural durability, and a low susceptibility to corrosion and fouling, the following flame arrester devices have been found effective in preventing flashback flames in an open environment from entering vent openings of a cargotank containing typical bulk fuels: (1) single 30-mesh stainless-steel screen, (2) dual 20-mesh stainless-steel screen, (3) spiral-wound, crimped stainless steel ribbon, and (4) packed bed of aluminum Ballast rings with single 30-mesh stainless-steel screen retainers. Ethylene, which is a gas at ambient temperature and pressure, is not a typical bulk cargo fuel.
- (2) Based upon the ability to withstand 30 minutes of continuous burning of a propane/air mixture, the following flame arrester devices have been found effective in sustaining the flame from typical bulk cargo fuels:
 (1) single 30-mesh stainless-steel screen, (2) dual 20-mesh stainless-steel screen, (3) spiral-wound, crimped stainless-steel ribbon, and (4) packed bed of Ballast rings with single 30-mesh stainless steel screen retainers. Spiral-wound, crimped metal ribbon arresters appear to have a finite time duration for sustained burning conditions, and should therefore be evaluated for the specific fuel and at the most severe condition of the intended applications. None of the flame arrester devices tested is effective in sustaining the flame from an ethylene/air mixture for 30 minutes duration.
- (3) Based upon the inverse relationship between the equilibrium temperature of a screen flame arrester at sustained burning conditions and the fuel/air mixture flowthrough velocity, it is recommended that in fuel transfer operations the rate of fuel flow should be fast enough to keep the exhaust velocity of vented flammable mixture well above the laminar burning velocity of the fuel being transferred. In the event of a flashback flame, this safety precaution will aid in keeping the screen flame arrester on the vent from over-heating by a sustained flame.
- (4) The selection of a location for the flame arrester device on the vent stack should be limited to the very end of the pipe. The flame quenching ability of the arrester is reduced by any length of pipe, housing, or mechanical device downstream of the arrester. Screen-type flame arresters are effective only if they are undamaged by punctures or tears in the wire mesh and there are no gaps or holes around the periphery larger than the openings specified for the 20- or 30-mesh screen. All flame arrester devices should be periodically inspected for damage and cleaned to remove fouling and corrosion.

(5) The selection of materials used in the construction of arresters should be based on their compatibility with the local environment and the fuel vapors to be encountered. However, stainless steel is recommended.

The data and experience obtained from these flashback flame and sustained burning tests is limited to those fuel and air mixtures tested in a 15.2-cm-(16-in.+) diameter pipe size. It is recommended that extrapolation of this data should be limited to the following:

- (1) Application to other fuels should be limited to those hydrocarbon fuels that have similar combustion characteristics to those fuels tested.
- (2) Applications scaled down to pipe sizes smaller than 15.2-cm (6-in.) diameter are considered to be conservative.
- (3) Scaled-up applications should be limited to pipe sizes no larger than a 20.3-cm (8-in.) diameter, providing adequate consideration is given to structural strength.

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APPENDIX A
TEST CONFIGURATION LOG

Configuration	n Test No.	Description
100 to 112	1488 to 1495	The first thirteen test configurations were evolved during the facility checkout tests. They included the preliminary installation of a subscale flame chamber that was later replaced by the full-scale flame chamber and the exhaust collector burn stack. Flame sensors on the flame chamber outer wall were repositioned from the horizontal center line to the top center line. Three igniter positions used in the flame chamber were (1) upstream, (2) middle, and (3) downstream. An aluminum flame shield was installed on the inlet piping upstream of the flame arrester test section. Also, a second aluminum flame shield was installed in front of the downstream flame chamber frangible diaphram. Fuels used on these checkout tests were gasoline and commercial grade propane. The test arresters included both the dual 20-mesh screens and the single 30-mesh screen.
113	1496 (A-C)	This test configuration is shown in Figure 7-2. Flame arrester: dual 20-mesh screens Fuel: propane Igniter position: upstream
114	1497 (A-C)	Flame arrester: dual 20-mesh screens Fuel: propane Igniter position: downstream
115	1498 (A-D)	Flame arrester: single 30-mesh screen Fuel: propane Igniter position: downstream
116	1499 (A-C)	Flame arrester: single 30-mesh screen Fuel: propane Igniter position: upstream
117	1500 (A-C)	Flame arrester: single 30-mesh screen Fuel: ethylene Igniter position: upstream
118	1501 (A)	Changed the exhaust collector burn-stack flame arrester from an Amal spiral-wound, crimped stain-less-steel ribbon to a Shand and Jurs spiral-wound, crimped aluminum ribbon assembly. Flame arrester: single 30-mesh screen Fuel: ethylene Igniter position: upstream

nfligaretion	Test No.		Description
119	1501 (B-D)	Flame arrester: Fuel: Igniter position:	single 30-mesh screen ethylene downstream
159	1502 (A)	Flame arrester: Fuel: Igniter position:	none ethylene downstream
101	1502 (B-D)	Flame arrester: Fuel: Igniter position:	dual 20-mesh screens ethylene downstream
122	1503 (A-C)	Flame arrester: Fuel: Igniter position:	dual 20-mesh screens ethylene upstream
123	1504 (A-C)	Flame arrester: Fuel: Igniter position:	crimped ribbon propane upstream
NOTE:		llowing tests were e upstream position	made with the igniter unless otherwise
124	1505 (A-D)	Flame arrester: Fuel:	crimped ribbon ethylene
125	1506 (A-D)	Flame arrester: Fuel:	crimped ribbon gasoline
106	1507 (A-D)	Flame arrester: Fuel:	none gasoline
127	1507 (C) 1508 (A-B)	Flame arrester: Fuel:	single 30-mesh screen gasoline
Rish	1508 (C-E)	Flame arrester: Fuel:	dual 20-mesh screens gasoline
109	1509 (A-C)	Flame arrester: Fuel:	packed bed of rings gasoline
130	1510 (A-C)	Flame arrester: Fuel:	packed bed of rings with single 30-mesh screen gasoline

Configuration No.	Test No.		Description
131	1511 (A-D)	Flame arrester: Fuel:	packed bed of rings with single 30-mesh screen ethylene
132	1512 (A-C)	Flame arrester: Fuel:	packed bed of rings with single 30-mesh screen propane
133	1513 (A-C)	Flame arrester: Fuel:	single 30-mesh screen methyl alcohol
134	1513 (D)	Flame arrester: Fuel:	none methyl alcohol
135	1514 (A-C)	Flame arrester: Fuel:	dual 20-mesh screens methyl alcohol
136	1515 (A-C)	Flame arrester: Fuel:	dual 20-mesh screens toluene
137	1515 (D)	Flame arrester: Fuel:	none toluene
138	1516 (A-D)	Flame arrester: Fuel:	single 30-mesh screen toluene
139	1517 (A-C)	Flame arrester: Fuel:	single 30-mesh screen diethyl ether
140	1517 (D)	Flame arrester: Fuel:	none diethyl ether
11+1	1518 (A-C)	Flame arrester: Fuel:	dual 20-mesh screens diethyl ether
142	1519 (A-D)	Flame arrester: Fuel:	dual 20-mesh screens butane
143	1519 (E)	Flame arrester: Fuel:	none butane
144	1520 (A-C)	Flame arrester: Fuel:	single 30-mesh screen butane
145	1521 (A-C)	Flame arrester: Fuel:	single 30-mesh screen acetaldehyde

Configuration	Test No.		Description
146	1521 (D)	Flame arrester: Fuel:	none acetaldehyde
147	1522 (A-C)	Flame arrester: Fuel:	dual 20-mesh screens acetaldehyde
148	1523 (A-B)	Changed the test a test configuration Flame arrester: Fuel:	assembly to the sustained burning on. crimped ribbon propane
149	1524 (A)		ocouples in the test arrester rounded to closed-end grounded. crimped ribbon propane
150	1524 (B-C)	Flame arrester: Fuel:	crimped ribbon ethylene
151	1525 (A)	Flame arrester: Fuel:	packed bed of rings with single 30-mesh screen propane
152	1525 (B - C)	Flame arrester: Fuel:	packed bed of rings with single 30-mesh screen ethylene
153	1526 (A)	Flame arrester: Fuel:	15.2-cm- (6.0-in) diameter single 30-mesh screen propane
154	1526 (в)	Flame arrester: Fuel:	15.2-cm- (6.0-in) diameter dual 20-mesh screens propane
195	1527 (A)	Flame arrester: Fuel:	25.4-cm- (10.0-in) diameter single 30-mesh screen propane
156	1527 (В)	Flame arrester: Fuel:	25.4-cm- (10.0-in) diameter dual 20-mesh screens propane

APPENDIX B

TABULAR SUMMARY OF STEADY-STATE MEASURED AIR AND FUEL SYSTEM TEST CONDITIONS

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TABULAR SUMMARY OF TRANSIENT-STATE MEASURED FLAME SPEED AND PEAK PRESSURE RISE

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APPENDIX D

TABULAR SUMMARY OF AVERAGED MEASURED FLAME SPEED AND PEAK PRESSURE RISE FOR FUELS

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APPENDIX E

TABULAR SUMMARY OF TEMPERATURE MEASUREMENTS FOR SUSTAINED BURNING TESTS

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